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AERODYNAMICS, KINEMATICS, AND ECOLOGY  
OF SLOW FLIGHT IN BIRDS

By

KRISTEN ELIZABETH CRANDELL

B.A. Lewis & Clark College, Portland, OR, 2009

Dissertation

presented in partial fulfillment of the requirements  
for the degree of

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Approved by:

Sandy Ross, Dean of The Graduate School  
Graduate School

Bret Tobalske, Chair  
Division of Biological Sciences

Douglas Emlen  
Division of Biological Sciences

Art Woods  
Division of Biological Sciences

Andrew Biewener  
Department of Organismic and Evolutionary Biology  
Harvard University

Sharon Swartz  
Department of Ecology and Evolutionary Biology  
Brown University

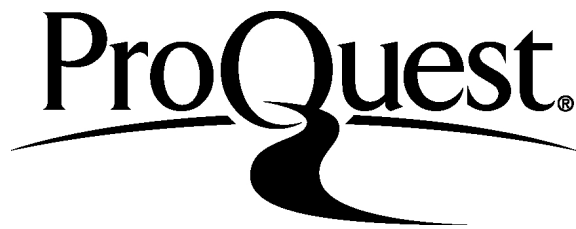
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## Aerodynamics, kinematics, and ecology of slow flight in birds

Chairperson: Bret W. Tobalske

The overarching goal of my dissertation is to elucidate the force production mechanisms of slow flight in birds. Slow flight is extremely energetically costly per unit time, yet highly important for takeoff and survival in birds. However, at slow speeds it is presently thought that most birds do not produce beneficial aerodynamic forces during the entire wingbeat: instead, they fold or flex their wings during upstroke, prompting the long-standing prediction that the upstroke produces trivial forces. Here, I examined the kinematics, aerodynamics, skeletal drivers, and potential ecological influences of force production in flight. In chapter one, I establish that wings in upstroke posture are capable of producing beneficial aerodynamic forces. Chapter two illustrates diamond doves that keep their wings extended in a “wingtip-reversal” upstroke (at  $Re=50,000$ ) produce a kinematic and aerodynamic signature similar to the clap-and-peel mechanism previously reported only in insects ( $Re=8,000$ ). In contrast, zebra finch use a “flexed-wing” upstroke that is aerodynamically inactive. Integrating an XROMM (X-ray Reconstruction of Moving Morphology) study of pigeons and starlings in chapter three, I demonstrate that both upstroke styles have similar skeletal kinematics (with a few notable exceptions), but the timing and extent of motion differs. Lastly, chapter four faces birds with an ecologically relevant task: transitioning from a compliant substrate reduces initial flight velocities, and birds do not appear to modulate force production to compensate. Collectively, I reveal that the clap and fling mechanism utilized by many species is a wing motion that is aerodynamically beneficial and largely due to an interaction of the skeletal elements. These four chapters illuminate an energetically costly and ecologically relevant period of flight.

## **Aerodynamics, kinematics, and ecology of slow flight in birds**

To initiate and end locomotion, an animal must accelerate from rest and decelerate to stop. Most studies of movement have focused on steady-state locomotion, during which the organism moves at a constant pace (Dickinson et al., 2000). In the natural world, this type of locomotion may be the exception to the norm; animals are constantly starting and stopping. For many organisms, these transitional phases may be the most crucial: to escape predation, impress a potential mate, and move between resources. Thus, the ability to change velocity is as important as the ability to maintain steady velocity in the animal world.

Acceleration (and steady-state locomotion) in terrestrial environments can be conveniently divided into two parts for vertebrates: the stance and swing phases. During the stance phase, the limb directly applies a force to the environment (i.e., the ground). The swing phase occurs as the limb is lifted off the ground and cycled forward to prepare for the next stance phase. Unsurprisingly, this phase is not directly producing any useful forces, because it is not in contact with the ground. In contrast, animals in a fluid environment (air or water) are constantly in direct contact with their environment. This continuous contact may provide an opportunity to produce useful forces during the traditional terrestrial swing phase.

Birds are able to capitalize on the swing phase, the upstroke, during fast flight: they produce a force to offset bodyweight (lift). In contrast, the functional significance of the upstroke during slow flight is unknown. The critical difference between these two speeds has to do with a continuous property of air: as the speed of the object in air increases, so too does the number of air molecules with which the

object (such as a wing) interacts. Thus, during slow flight a majority of the aerodynamic power required to fly is expended to support the animal's weight, while during fast flight power requirements for weight support (known as induced power) are much lower. This implies that a bird is operating near its aerodynamic limits during slow flight. Even so, slow flight is a crucial phase for a bird. In order to fly from a standstill, a bird must accelerate through a slow phase to reach a comfortable cruising speed. Given that slow, accelerating flight is energetically costly, near aerodynamic limits, and likely under selective pressures for brevity, I predict that birds should use mechanisms to rapidly accelerate following takeoff. One such method a bird may be able to accomplish this rapid acceleration, without an added energetic cost, may be during upstroke.

During slow flight, birds produce one of two upstroke styles: the wingtip-reversal upstroke, wherein the wing is extended and the hand wing goes through significant supination and pronation, or the flexed-wing upstroke, wherein the wing is flexed toward the body, and extended to initiate the next downstroke. It was first proposed by Marey in 1890 (Marey, 1890) that the tip-reversal upstroke may be aerodynamically active. Various studies hence have substantiated this claim (Brown, 1953; Hedrick and Biewener, 2007; Lorenz, 1933; Ros et al., 2011) yet the aerodynamic and internal mechanism remains unknown.

The overarching goal of my dissertation is to elucidate the force production mechanisms of slow flight in birds. Herein, I address the aerodynamics, kinematics, skeletal drivers, and potential ecological influences of force production in flight. The first chapter models the aerodynamic force production of a wing in upstroke and

downstroke posture, indicating that the posture of tip-reversal upstroke is capable of producing significant aerodynamic forces. The second chapter examines force production *in vivo*. I establish that the tip-reversal upstroke capitalizes on a time-dependent “clap and peel” mechanism, wherein the left and right wings interact at the upstroke to downstroke transition to produce additional circulation greater than that which is generated in a non clap-and-peel transition. In contrast, the flexed wing upstroke produces minimal induced momentum to the wake, and is largely aerodynamically inactive. The third chapter examines the underlying mechanical driver of these mechanisms via an XROMM (X-ray Reconstruction of Moving Morphology) reconstruction of skeletal kinematics. Here, I establish that both upstroke styles have generally similar skeletal kinematics (with a few notable exceptions), but the timing and extent of each motion differs to produce drastically different external wing motion. Lastly, I explore the ability of birds to modulate the forces and kinematics observed in the first three chapters when faced with an ecologically-relevant task. Transitioning from a compliant substrate reduces initial flight velocity, and birds do not appear to modulate force production to compensate. Together, these four chapters help illuminate an energetically costly and ecologically relevant period of flight.

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## RESEARCH ARTICLE

# Aerodynamics of tip-reversal upstroke in a revolving pigeon wing

Kristen E. Crandell\* and Bret W. Tobalske

Field Research Station at Fort Missoula, Division of Biological Sciences, University of Montana, Missoula, MT 59812, USA

\*Author for correspondence (kristen.crandell@umontana.edu)

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### SUMMARY

**During slow flight, bird species vary in their upstroke kinematics using either a ‘flexed wing’ or a distally supinated ‘tip-reversal’ upstroke. Two hypotheses have been presented concerning the function of the tip-reversal upstroke. The first is that this behavior is aerodynamically inactive and serves to minimize drag. The second is that the tip-reversal upstroke is capable of producing significant aerodynamic forces. Here, we explored the aerodynamic capabilities of the tip-reversal upstroke using a well-established propeller method. Rock dove (*Columba livia*,  $N=3$ ) wings were spread and dried in postures characteristic of either mid-upstroke or mid-downstroke and spun at *in vivo* Reynolds numbers to simulate forces experienced during slow flight. We compared 3D wing shape for the propeller and *in vivo* kinematics, and found reasonable kinematic agreement between methods (mean differences 6.4% of wing length). We found that the wing in the upstroke posture is capable of producing substantial aerodynamic forces. At *in vivo* angles of attack (66 deg at mid-upstroke, 46 deg at mid-downstroke), the upstroke wings averaged for three birds produced a lift-to-drag ratio of 0.91, and the downstroke wings produced a lift-to-drag ratio of 3.33. Peak lift-to-drag ratio was 2.5 for upstroke and 6.3 for downstroke. Our estimates of total force production during each half-stroke suggest that downstroke produces a force that supports 115% of bodyweight, and during upstroke a forward-directed force (thrust) is produced at 36% of body weight.**

Key words: upstroke, propeller model, pigeon, *Columba livia*, flight, wing, lift, drag, tip-reversal.

### INTRODUCTION

In almost all birds flight plays a crucial role in survival, whether in foraging, attracting mates, defending territories, migrating or avoiding predators. In order to accomplish all of these tasks in flight, birds must first take off and accelerate to the appropriate speed. During this acceleration, flying birds appear to have two unique patterns of wing kinematics, associated with wing morphology. The first group fully flexes their wings on the upstroke and has low aspect ratio, rounded wings. The second group uses a ‘tip-reversal’ upstroke, in which the distal wing is heavily supinated, and includes birds with high aspect ratio, pointed wingtips (Brown, 1963; Tobalske, 2000; Tobalske et al., 2003). A notable exception to this pattern is seen in a group of birds designed for explosive take-off, *Phasianidae*, which have rounded, low aspect ratio wings but a prominent tip-reversal upstroke (Tobalske and Dial, 2000).

The functional significance of the tip-reversal upstroke has been debated for nearly a century. Early flow-visualization evidence suggested that the tip-reversal upstroke is aerodynamically inactive, meaning that it does not produce lift for weight support or thrust (Spedding et al., 1984; Spedding, 1986). However, this is unlikely given dramatic differences in wing kinematics (Brown, 1953; Tobalske and Dial, 1996) between upstroke styles, and observed increases in body acceleration during the upstroke (Brown, 1963; Aldridge, 1987). These early measures of body acceleration do not account for wing inertia in calculating center of mass acceleration, and thus may not accurately reflect performance. Recent evidence for an active upstroke is found in feather loading during the upstroke, indicating a net force directed anteriorly to the bird (Corning and Biewener, 1998), and significant differential pressure on the surfaces of the wing during upstroke through take-off and landing

(Usherwood et al., 2005). Similarly, an analysis of instantaneous force changes as measured by accelerometers indicates that the tip-reversal upstroke in a cockatiel (*Nymphicus hollandicus*) provides 14% the lift of downstroke (Hedrick et al., 2004). If the upstroke provides a portion of the lift or thrust forces necessary for flight, birds that exhibit a tip-reversal upstroke may have a large advantage during slow flight in comparison to birds that flex their wings on the upstroke. Slow flight is energetically costly (Rayner, 1995); induced drag is high, so much of the power required to fly slowly is used to support the animal’s weight (Ellington, 1991). Thus, we expect that animals should capitalize on mechanisms that allow them to rapidly accelerate to speeds that require less power output. Therefore, we predict that a tip-reversal upstroke will be aerodynamically active.

Here, we tested for an aerodynamic function of the upstroke in the rock dove (*Columba livia*, Gmelin 1789, hereafter ‘pigeon’) using a well-established propeller method (Usherwood and Ellington, 2002a; Usherwood and Ellington, 2002b) that has thus far only been used to model wing function during mid-downstroke (Usherwood and Ellington, 2002a; Usherwood and Ellington, 2002b; Altshuler et al., 2004; Usherwood, 2009). We used *in vivo* kinematics to model the posture of spread wings and to determine the Reynolds number estimated from wingtip angular velocity. We then directly measured lift and drag by mounting the dried-wing propellers on a force plate.

### MATERIALS AND METHODS

#### Kinematics

A captive-bred pigeon was housed in an outdoor aviary at the Field Research Station at Fort Missoula. The bird was trained to fly

between horizontal platforms positioned approximately 2 m apart. Prior to kinematic recordings, the bird was marked with black permanent marker at the shoulder, wrist and each primary feather tip, and at three locations on the ventral surface of the body for later digitization. All care and experimental procedures were approved by the University of Montana IACUC.

We recorded the flight of the pigeon using four high-speed cameras: one Photron SA-3 (Photron USA Inc., San Diego, CA, USA), two Photron 1024 PCI and one Fastec Troubleshooter (Fastec Imaging, San Diego, CA, USA), synchronized using a transistor–transistor logic (TTL)-pulse. One camera was placed dorsally and behind the flight path, two were ventrally placed behind and in front of the flight path, and a fourth was placed horizontal to the bird in front of the flight path. Video recordings were made at 1000 Hz with a shutter speed of 1/10,000 s.

To analyze the kinematic data, we recorded a 36-point calibration volume (approximately 30×45×80 cm) for the space between the platforms. A direct linear transformation converted the four camera views into 3D coordinates (MATLAB R2010a, The MathWorks Inc., Natick, MA, USA) (Hedrick, 2008). Markers along the wings were digitized from recorded images to obtain 3D wing kinematics. The third wingbeat cycle following take-off was used.

We performed subsequent analysis using custom scripts in MATLAB. Angular velocity ( $\text{rad s}^{-1}$ ) of the hand wing was calculated by determining a central position in three dimensions between the wrist, the 10th primary feather and the 1st secondary feather, to represent the total hand wing. We then determined the velocity vector of the hand wing, relative to the shoulder of the bird. To determine the angle of attack ( $\alpha$ , deg) in global coordinates (accounting for translation), we measured the angle between the velocity vector of the hand wing and the vector normal to the hand-wing plane, and subtracted this from 90 deg.

### Morphology

Wings were removed from three deceased birds (body mass, 446±69 g) and dried in a low temperature oven at 40°C for 1 week. We obtained paired models for each individual, with one wing in mid-upstroke posture and the other in mid-downstroke posture. Each wing was mounted on a brass rod at the proximal edge of the head of the humerus to create a one-blade propeller, with the rod acting as a counterweight (Fig. 1). Wings were photographed and analyzed in ImageJ (v1.43u, National Institutes of Health, Bethesda, MA, USA) and MATLAB to determine wing length and the first, second and third moments of area [following Usherwood and Ellington (Usherwood and Ellington, 2002b)] (Table 1). The second moment of area directly correlates with aerodynamic force, and the third moment with torque (Weis-Fogh, 1973). Thus, these two measurements allow us to account for slight differences in wing shape between birds when calculating mean vertical and horizontal forces (Eqns 1 and 2).

### Propeller

Each wing was spun once at each angle of attack ( $\alpha$ ), in increments of approximately 10 deg from –10 to 90 deg. Wings were mounted to a brushless DC motor (NEMA 34 motor, Anaheim Automation Inc., Anaheim, CA, USA), and controlled with an Anaheim Automation driver and power supply (DCL 601USB, MBC 12101 and PSA 40V8A), with SMC60WIN software (v. 2.01). Revolutions per minute (r.p.m.) values were based on the mean angular velocity of the wingtip during the middle third of each half-stroke, as determined from *in vivo* kinematics for upstroke and downstroke. The third wingbeat following take-off of the live pigeon had an

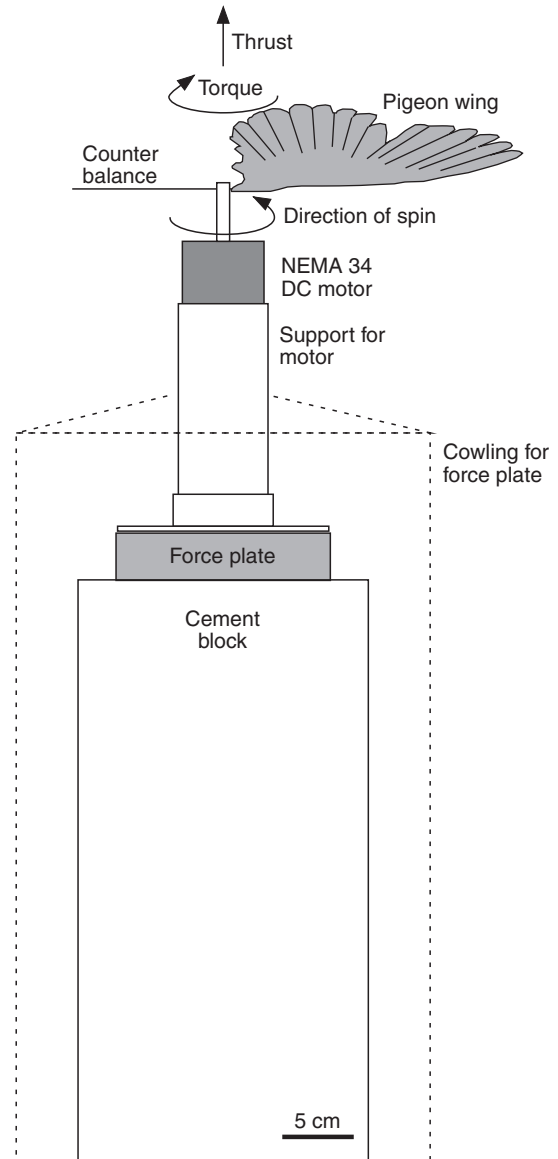


Fig. 1. The propeller setup.

angular velocity of  $69 \text{ rad s}^{-1}$  for the mid-third of downstroke, and  $96 \text{ rad s}^{-1}$  for the mid-third of upstroke, which translated into 376 and 572 r.p.m., respectively.

The propeller shaft was constructed to allow the insertion and manipulation of the brass rod for easy rotation of the wing over a full range of  $\alpha$ . We measured  $\alpha$  as the acute angle between horizontal and a line described by two points on the wing: midway between the wrist, tip of the 10th primary feather and the tip of the 1st primary feather. We visualized these points using a digital video camera (Photron SA-3 using PFV 3.1.8 software) that was placed horizontal to the wing. We measured  $\alpha$  before spinning the wing and during rotation, both to account for deformation due to aerodynamic loading of the wing and to report active  $\alpha$  for final analysis. Active  $\alpha$  was on average  $2 \pm 3$  deg less than static  $\alpha$ .

The wings were mounted so that induced velocities in the wake would generally be directed upwards, away from the force plate. For the downstroke posture, for example, this meant the wing was mounted with its ventral surface facing upwards when  $\alpha > 0$  deg. For

upstroke posture wings, the secondaries contacted the motor when  $\alpha < 50$  deg, so we oriented the wing so that the ventral surface faced downwards. The aerodynamic contributions of the counterweight were accounted for by spinning a rod with double the length of the counterweight, centered on the propeller shaft, at appropriate r.p.m. The resulting measurements were halved and subtracted from observed wing values.

We used a custom-built force plate (15×15 cm, Bertec Corp., Columbus, OH, USA) to measure vertical force along the  $z$ -axis and torque about the  $z$ -axis due to drag (Usherwood, 2009). The force plate featured a known conversion of  $10,000 \text{ mN V}^{-1}$  for force and  $800 \text{ mN V}^{-1}$  for torque. Voltage output from the force plate was converted and amplified with a digital gain of  $5\times$  using a Bertec model M6810 amplifier. We imported these data to a computer using an ADInstruments PowerLab 8SP A/D converter sampling at 1000 Hz and Chart v5.2 software using a 1 Hz low-pass digital filter (ADInstruments Inc., Colorado Springs, CO, USA). A cardboard cowling (shield) surrounded the base of the motor and the force plate in order to isolate the force plate from air velocities induced by the spinning wings (Fig. 1).

#### Wing shape during spinning versus flight

To compare deflection of primary feathers during model rotation with *in vivo* wing shape, we used 3D video recording and analysis on the propeller model (see ‘Kinematics’, above). We videotaped the upstroke and downstroke wing models when spinning with  $\alpha$  comparable to *in vivo* measurements (66 deg for upstroke, 46 deg for downstroke wings). To compare wing shape between the live bird and propeller, digitized points were standardized to a right-hand rule, local coordinate system rooted at the wrist, with the  $x$ -axis extending through the 10th primary feather. Conversion from global to local coordinates was accomplished using a series of Euler angle transformations.

We assessed variation for each feather between a model wing (Fig. 2, blue lines) and live animal (red lines) in both the proximal–distal and ventral–dorsal axes. Feather offset was measured as the linear distance in three dimensions from each primary feather tip to its corresponding feather in the live bird, and normalized by dividing by the wing length (Fig. 2). Feathers in upstroke-postured wings differed from *in vivo* feathers by  $1.96 \pm 0.97 \text{ cm}$  or  $6.05 \pm 2.75\%$  of wing length. Downstroke posture wings differed by  $2.02 \pm 0.91 \text{ cm}$  or  $6.78 \pm 3.06\%$  of wing length.

#### Determining coefficients of lift and drag

Data from the force plate were converted to the coefficient of lift or drag by a series of equations presented by Usherwood and Ellington (Usherwood and Ellington, 2002a). The mean vertical force reported by the force plate ( $F_v$ , mN) was converted to the mean vertical force coefficient ( $C_v$ ):

$$C_v = \frac{2F_v}{\rho S_2 \Omega^2}, \quad (1)$$

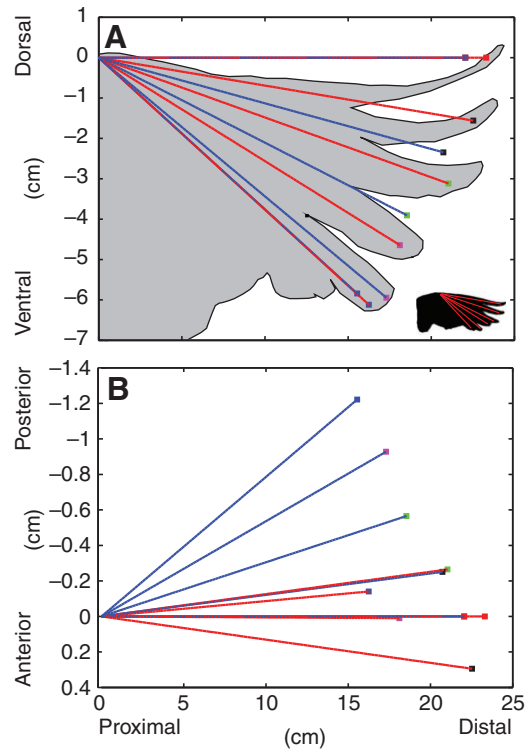


Fig. 2. A comparison of *in vivo* and propeller wing feather deflection at a 66 deg angle of attack for a wing during mid-upstroke. Live bird and propeller data were normalized to a standard coordinate system along the 10th primary feather. Each line represents a position vector from the wrist base to the tip of each primary feather (colored squares). The live bird is shown in red, the upstroke wing model from one bird in blue. (A) Proximal–distal and ventral–dorsal axis, with feather outlines in gray. (B) Proximal–distal and anterior–posterior axis.

where  $\rho$  is the air density ( $1.07 \text{ kg m}^{-3}$ , in Missoula, MT, USA) during the experiments,  $S_2$  is the second moment of area (Table 1) and  $\Omega$  is the angular velocity ( $\text{rad s}^{-1}$ ). Then, we converted the mean torque ( $Q$ , mN m) about the  $z$ -axis of the plate to a horizontal force coefficient ( $C_h$ ):

$$C_h = \frac{2Q}{\rho S_3 \Omega^2}, \quad (2)$$

where  $S_3$  is the third moment of area.

Using both of these values, we determined the dimensionless coefficient of lift ( $C_L$ ) and drag ( $C_D$ ), using a value for the local downwash angle ( $\epsilon$ ) determined geometrically assuming a triangular distribution of induced velocity along the wing (Usherwood and Ellington, 2002a). This distribution has been confirmed for the

Table 1. Morphometrics for pigeon (*Columba livia*) wings used for propeller model ( $N=3$ )

	Downstroke	Upstroke
Wing length ( $R$ ; m)	$3.14\text{E}-01 \pm 2.47\text{E}-02$	$2.43\text{E}-01 \pm 1.09\text{E}-02$
Wing Area ( $S$ ; $\text{m}^2$ )	$3.35\text{E}-02 \pm 4.02\text{E}-03$	$2.05\text{E}-02 \pm 2.92\text{E}-03$
Second moment of wing area ( $S_2$ ; $\text{m}^4$ )	$8.37\text{E}-04 \pm 3.22\text{E}-04$	$2.36\text{E}-04 \pm 2.21\text{E}-05$
Third moment of wing area ( $S_3$ ; $\text{m}^5$ )	$1.76\text{E}-04 \pm 8.07\text{E}-05$	$3.54\text{E}-05 \pm 3.66\text{E}-06$

For each bird, the right wing was positioned in upstroke posture, the left in downstroke.



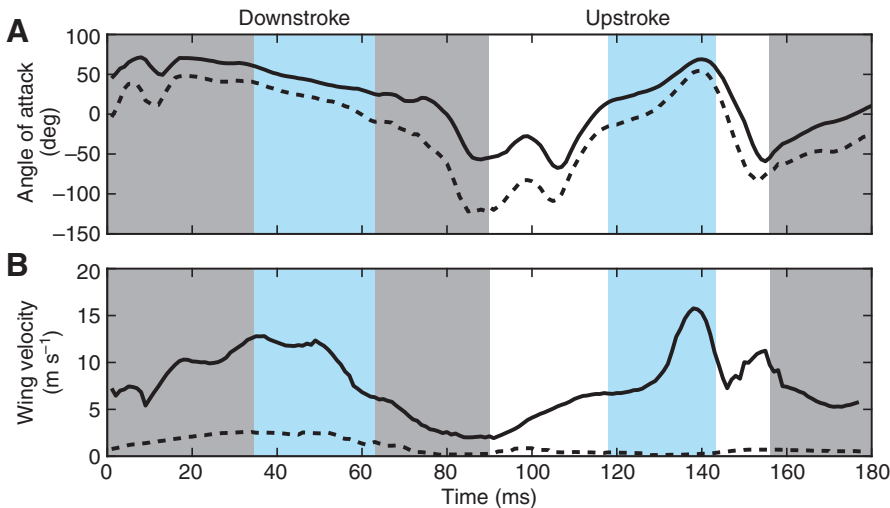


Fig. 3. Kinematic parameters extracted from the third wingbeat cycle; downstroke is shaded, and times approximated by the propeller model are highlighted in blue. (A) Geometric angle of attack ( $\alpha$ , deg):  $\alpha$  is shown by the solid line,  $\alpha$  accounting for the estimated induced velocity is shown by the dashed line (see Discussion). (B) Wing velocity ( $\text{m s}^{-1}$ ) for a full wingbeat cycle: wingtip is shown by the solid line, wrist is shown by the dashed line.

propeller model with chukar partridge (*Alectoris chukar*) wings using particle image velocimetry (PIV) (Heers et al., 2011). Thus,  $C_L$  was:

$$C_L = (C_v \cos \epsilon + C_h \sin \epsilon) (\cos \epsilon)^{-2} \quad (3)$$

and  $C_D$  was:

$$C_D = (C_h \cos \epsilon - C_v \sin \epsilon) (\cos \epsilon)^{-2}. \quad (4)$$

While *in vivo* upstroke kinematics qualitatively differ from downstroke kinematics, the tip-reversal upstroke rotates around the wrist, where the pigeon averages an angular velocity of only  $6.5 \text{ rad s}^{-1}$ . Thus, by positioning the wrist over the axis of rotation in our propeller model, we can assume a constant angular velocity along the hand wing during upstroke, and use the above calculations for both upstroke and downstroke wings.

#### Estimating *in vivo* performance

Combining *in vivo* kinematics with force coefficients extracted from the propeller model allows us to consider the force production of a pigeon during the third wingbeat cycle following take-off using coefficients measured with the propeller. First, assuming a hovering case, we can calculate lift (**L**) as:

$$\mathbf{L} = \frac{1}{2} \rho V^2 C_L S_2 \quad (5)$$

and drag (**D**) as:

$$\mathbf{D} = \frac{1}{2} \rho V^2 C_D S_2; \quad (6)$$

then, the resultant force (**F<sub>R</sub>**) is calculated as:

$$\mathbf{F}_R = \sqrt{(\mathbf{L}^2 + \mathbf{D}^2)}. \quad (7)$$

However, because the bird is moving forward, we must account for both translational velocity (Dickinson and Dickinson, 2004) and  $\Omega$  along the wing ( $\Omega_r$ ). We used a blade element analysis (Osborne, 1951) and expanded Eqn 1 to find the vertical force (**F<sub>v</sub>**):

$$\mathbf{F}_v = \rho C_v \int_{r=0}^R (\Omega_r r + V_T)^2 c_t dr, \quad (8)$$

where  $c$  is the chord length and  $V_T$  is the translational velocity of the wing accounting for stroke plane angle. We estimated  $\Omega_r$  directly from kinematic data by dividing the leading edge (defined by a line between shoulder, wrist, and wingtip) into equal segments and taking the  $\Omega_r$  for mean mid-stroke. Wing velocity increased toward the distal wingtip (Fig. 3B, wrist and wingtip). For upstroke, these

calculations considered only the hand wing, as the proximal wing is flexed near the body with a low velocity (Fig. 3B, dashed line), and thus was assumed to be aerodynamically inactive. We substituted  $C_h$  for  $C_v$  in Eqn 8 to estimate **F<sub>h</sub>**. The resultant of these forces then estimates net force production.

#### Data analysis

We computed polars of mean  $C_L$  among the three birds sampled as a function of mean  $C_D$  as well as mean lift-to-drag ratio ( $C_L:C_D$ ) as a function of  $\alpha$ . To accomplish this, we first interpolated 95 points for  $C_L$  and  $C_D$ , to account for subtle differences in  $\alpha$  measured between wing samples. We calculated between  $\alpha = -10$  deg and  $\alpha = 85$  deg using IGOR Pro (v. 6.01, Wavemetrics, Inc., Beaverton, OR, USA). We then computed a mean and standard deviation among birds for each point in the interpolated series. Herein, we report these means  $\pm$  s.d.

## RESULTS

### Kinematics

The third wingbeat cycle following take-off was used in analyses. Angle of attack (Fig. 3A) ranged from 72 deg at the start of downstroke to  $-68$  deg during upstroke. Mean downstroke  $\alpha = 48$  deg, and during the mid-third of downstroke (Fig. 3A, blue) mean  $\alpha = 46$  deg. Mean upstroke  $\alpha = -2$  deg, but it ranged from  $-68$  to 71 deg. The mean  $\alpha$  during the mid-third of upstroke (Fig. 3A, blue) was  $\alpha = 66$  deg. Angular velocity ( $\Omega$ ) for the mid-third of downstroke was  $69 \text{ rad s}^{-1}$ , and for the mid-third of upstroke  $\Omega = 96 \text{ rad s}^{-1}$ . Tangential velocity of the wingtip varied throughout the stroke cycle, averaging  $10.1 \text{ m s}^{-1}$  for mid-downstroke and  $8.9 \text{ m s}^{-1}$  for upstroke (Fig. 3B). Maximum wingtip velocity for downstroke was  $12.8 \text{ m s}^{-1}$ ; for upstroke it was  $15.3 \text{ m s}^{-1}$ . Velocities along the wing measured from proximal to distal grew to a maximum at the wingtip (Fig. 3B, wrist and wingtip). However, during upstroke the proximal wing velocity remained low, likely due to being flexed against the body.

### Propeller force coefficients

Coefficients for lift and drag plotted against  $\alpha$  for all three birds revealed substantial lift production by the upstroke wings, with a general pattern that followed that of downstroke (Fig. 4). Upstroke-posture wings had a maximum  $C_L$  of  $1.42 \pm 0.08$  at  $\alpha = 53.8$  deg (Fig. 4A), and downstroke wings had a maximum  $C_L$  of  $1.77 \pm 0.43$  at  $\alpha = 60.8$  deg (Fig. 4C). For upstroke wings, the minimum mean

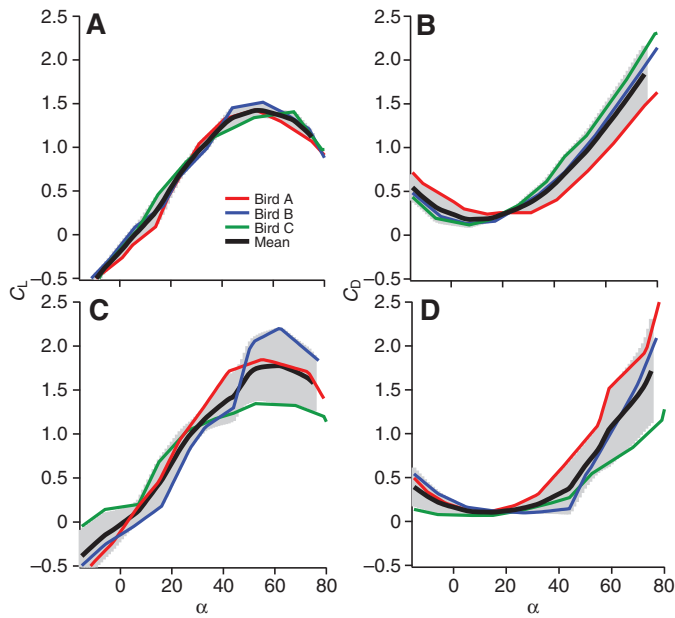


Fig. 4. Aerodynamic force coefficients for wing in upstroke (A,B) and downstroke (C,D) postures with reference to angle of attack ( $\alpha$ ).  $N=3$  birds. Mean among birds is shown by the bold black line, and the shaded area represents  $\pm$ s.d.

$C_D$  was  $0.18 \pm 0.09$  at  $\alpha=7.3$  deg (Fig. 4B) while for downstroke wings mean  $C_D$  was  $0.10 \pm 0.03$  at  $\alpha=14.3$  deg (Fig. 4D).

A diagram of  $C_L$  as a function of  $C_D$  revealed that the shape of the  $C_L:C_D$  polars was similar between wing postures (Fig. 5A). The difference in values appears to be caused by both a lower  $C_L$  and a higher  $C_D$  for upstroke wings, due to the minimum and maximum values seen in Fig. 4. The slopes of the  $C_L:C_D$  polar for upstroke and downstroke wings are almost identical (Fig. 5A), as evidenced by the similar  $\alpha$  at the maximum  $C_L:C_D$  (Fig. 5B). The maximum  $C_L:C_D$  for wings in downstroke posture was 6.3, which occurred at  $\alpha=26.5$  deg. Wings in upstroke posture had a lower maximum  $C_L:C_D$  of 2.5, at  $\alpha=28.0$  deg. At measured *in vivo*  $\alpha=66$  deg during mid-upstroke, the  $C_L:C_D$  is approximately 0.91. At this  $\alpha$ ,  $C_L=1.33$  and  $C_D=1.46$ . The  $C_L:C_D$  for downstroke posture was 3.33 at the *in vivo*  $\alpha=46$  deg, with  $C_L=1.54$  and  $C_D=0.49$ .

#### Bracketing *in vivo* performance

Various assumptions provided alternative estimates for *in vivo* force production. We observed minimum values for lift and drag if we assumed a bird was hovering, with no forward translational velocity (Eqns 5 and 6). In this instance, the bird (4.37 N) would produce a resultant force ( $F_R$ ) of 51.3% body weight (2.24 N) with both wings during mid-downstroke ( $\alpha=46$  deg). During mid-upstroke ( $\alpha=66$  deg), the pigeon could produce a  $F_R$  of 107% body weight (4.71 N).

However, these values likely do not represent *in vivo* performance: during the third wingbeat following take-off, the bird has already reached a forward velocity of  $2.9 \text{ m s}^{-1}$ . When including this translation in our estimation by incorporating translational velocity ( $V_T$ ) in a blade-element analysis (Eqn 8), mid-downstroke  $F_R$  produces 8.7 N, 198% of body weight. With a stroke plane angle of 38 deg during downstroke, the  $F_R$  is directed in front of the bird but near vertical, at 71 deg from the global horizontal plane, thus providing mostly weight support. During mid-upstroke,  $F_R=5.43$  N, 124% body weight. The upstroke stroke plane angle is 21 deg, and

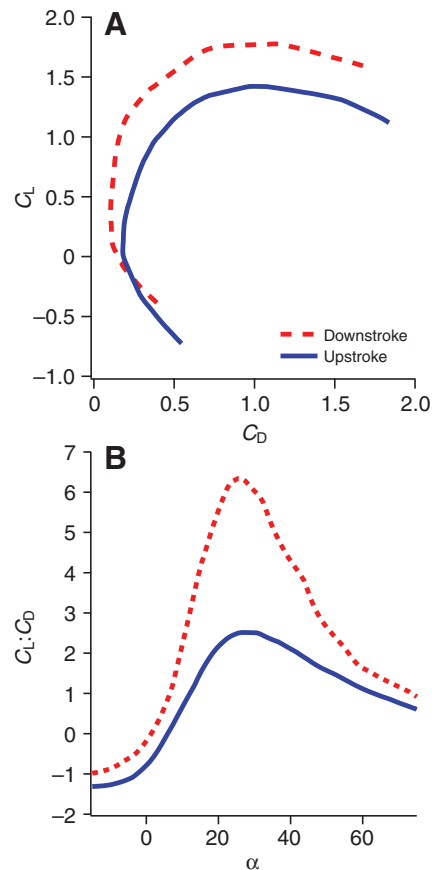


Fig. 5. (A) Polar diagram showing mean coefficient of lift ( $C_L$ ) as a function of mean coefficient of drag ( $C_D$ ) for  $N=3$  pigeon wings. (B) Mean lift-to-drag ratio ( $C_L:C_D$ ) against angle of attack ( $\alpha$ ) for  $N=3$  pigeon wings.

$F_R$  is oriented in front of the bird, 23 deg above the global horizontal plane, providing mostly thrust. These values probably represent extremes. Consider that mid-downstroke values appear to reflect the kinematics through the majority of downstroke (Fig. 3), while mid-upstroke kinematics reflects only a small portion of the entire half-cycle (Fig. 3). Extrapolating this calculation for the entire downstroke, which is approximately 58% of the wingbeat cycle, suggests that the bird produces a mean force of 115% body weight during downstroke. Throughout the entire upstroke, 42% of the wingbeat cycle, the mean force produced would be 36% bodyweight.

#### DISCUSSION

Given the debate over the functional significance of the tip-reversal upstroke (Brown, 1963; Spedding et al., 1984; Spedding, 1986; Aldridge, 1987; Hedrick et al., 2004; Tobalske et al., 2003), our most important result is that the model wings in upstroke posture produced a large amount of aerodynamic force (Figs 4 and 5). In fact, the downstroke and upstroke wings appear to perform similarly, both reaching maximum  $C_L:C_D$  at approximately  $\alpha=26$  deg (Fig. 5). This suggests that during mid-upstroke, significant force can be produced to help support weight or generate thrust, and the magnitude of this force is dependent on  $\alpha$  in much the same way as for downstroke.

At *in vivo* angles of attack, we estimate the upstroke is capable of producing mean force as great as 36% body weight during this half of the stroke cycle. This value is within the range of lift produced

by an active upstroke in hovering hummingbirds (33–48%) (Warrick et al., 2005; Warrick et al., 2009). Our results are supported by measures of feather bending during slow flight. Values of strain on primary feathers during upstroke are 43% that of downstroke (Corning and Biewener, 1998), suggesting that upstroke values reported here are not unreasonable for *in vivo* performance. The upstroke force we estimated also falls near the range of 14% the lift of downstroke measured with accelerometers in cockatiels (Hedrick et al., 2004).

Aerodynamically active upstrokes have been reported in several bat species that exhibit a similar wing-tip reversal pattern. The horseshoe bat (*Rhinolophus ferrumequinum*) has an angle of incidence at the wingtip during upstroke of 70 deg (Aldridge, 1986), close to our reported value of 66 deg. Introductory analyses of changes in acceleration found an aerodynamically active upstroke during vertical flight (Aldridge, 1987) and hovering (Aldridge, 1991) of bats. In-depth kinematic analyses accounting for inertial effects will be crucial to validate these findings. More recent analyses of the wake in slow-flying bats using PIV show a vortex ring shed at the end of upstroke (Hedenström et al., 2007; Johannson et al., 2008), indicating both thrust and weight support produced on the upstroke. Further PIV analyses show that the role of upstroke changes to only provide weight support at higher speeds in bats (Wolf et al., 2010). Our results suggest that birds exhibiting tip-reversal upstroke may use aerodynamic mechanisms similar to bats during slow flight.

Our findings indicate that slow-flying pigeons are using their wings at a lower  $C_L:C_D$  than is potentially available to them (Fig. 5A) during both upstroke and downstroke. However, they are operating at  $\alpha$  close to, or at, maximum lift performance (Fig. 5B). Peak  $C_L$  for downstroke was 1.8 at  $\alpha=61$  deg, and for upstroke it was 1.4 at  $\alpha=54$  deg. At such high  $\alpha$  on the upstroke, pigeons appear to be operating where lift and drag are almost equivalent ( $C_L:C_D=0.91$ ). The resultant force during mid-upstroke is directed forward, 23 deg from global horizontal, due in part to high drag. This suggests that drag may play an important role during the upstroke, possibly to assist in weight support or thrust. This may also be the case for bats in slow flight (Aldridge, 1987; Norberg, 1976), and for chukar partridges during wing-assisted incline running (Tobalske and Dial, 2007).

Our propeller measurements for downstroke were consistent with previously published values for a pigeon wing in downstroke posture. Usherwood reports a maximum  $C_L:C_D$  of 5.4 for a pigeon wing spun on a propeller emulating slow flight (Usherwood, 2009), in comparison to our  $C_L:C_D$  of 6.3 (Fig. 5B). Maximum  $C_L$  for the same wings was 1.64 (Usherwood, 2009), while our value was 1.77. Our slightly higher values are potentially due to differences in wing morphology (Table 1) and spinning speed. Estimated lift and drag values for downstroke from kinematic analyses of pigeons in slow flight (Berg and Biewener, 2008) yielded values of  $C_L=1.44\pm 0.29$  and  $C_D=1.01\pm 0.08$ , within the bounds of our values. Notably, the s.d. variation for  $C_L$  and  $C_D$  between wings increases with higher  $\alpha$  in downstroke wings, but equivalent variation did not occur for upstroke wings. During slow flight, unsteady aerodynamic effects likely dominate (Ellington, 1984). At such high  $\alpha$ , this may include periodic leading edge vorticity (Warrick et al., 2005) detachment. This flow separation may lead to unpredictable variation in both  $C_L$  and  $C_D$  throughout the half-stroke. In contrast, wings in upstroke posture, with smaller observed s.d. at high angles of attack, may reduce these effects *via* the separation of the primary feathers, allowing individual feathers to function as individual airfoils with lower  $\alpha$  (Brown, 1963). Further exploration using flow-visualization would help tease apart differences between *in vivo* and propeller aerodynamics.

It is important to note that values reported here represent values for mid-stroke. Our estimate of *in vivo*  $F_R$  provide a lower bracket for downstroke of 51% bodyweight during hovering, and an upper bracket of 198% when accounting for translation ( $V_T$ ). The resultants from these calculations are directed up 71 deg from global horizontal, indicating the downstroke acts to support weight. The same calculation for upstroke suggests a bracket between 107 and 124% body weight. The upstroke resultant is directed forward, 23 deg from global horizontal due to high levels of drag, suggesting upstroke provides thrust. These values likely provide a reasonable estimate for downstroke. Even then, wing velocity is greatest at mid-stroke, when our estimates are based, and resulting forces should be proportional to the square of wing velocity (Ellington, 1984). The values for upstroke are likely an over-estimate, as we applied static coefficients to a highly unsteady event. In the upstroke,  $\alpha$  and wing velocity are highly time variant (Fig. 3), and our estimation occurred at near peak wing velocity. Moreover, wing posture appears to change more dramatically throughout the upstroke when compared with the downstroke, and it may be the mid-stroke posture that is optimal for force production. Our kinematic analysis throughout the stroke cycle (Fig. 3) illustrates that as the hand wing pronates using a rapid ‘flick’ at the end of upstroke, a changing  $\alpha$  occurs, as also observed to a lesser extent in the downstroke in the pigeon (Warrick and Dial, 1998) and cockatiel (Hedrick et al., 2002). Such constant variation in wing velocity and  $\alpha$  in upstroke is also exhibited by hovering hummingbirds, and it appears that this helps to explain variation in circulation on the wing (Tobalske et al., 2007; Warrick et al., 2009). Thus, our final estimate for net force produced during both halves of the stroke cycle, using  $\Omega$ , and accounting for the percentage of the total wingbeat cycle, is probably the most accurate for predicting pigeon performance.

Additionally, it is critical to note that the dried wing models do not accurately reflect all aspects of mid-downstroke or mid-upstroke. The preparation process, in which wings are isolated and dried, likely changes the mechanical properties of the wing. Similarly, the posture adopted by the models may not be an accurate representation of actual kinematics, as evidenced in slight differences in feather location between pigeon and propeller measurements (Fig. 2). While our model reflects the forces during an impulse, in reality the wing is morphing throughout most of the stroke cycle. For example, wing flexibility and compliance throughout a stroke significantly increases lift in insects (Mountcastle and Daniel, 2009; Young et al., 2009). During the stroke cycle of the pigeon, significant pronation and supination also occur. This results in wing camber and wing twist, which significantly improve aerodynamic forces in the locust (Young et al., 2009). The propeller model wings are probably unable to fully reflect these characteristics.

Our measure of  $\alpha$  on the propeller is the geometric angle of attack, which does not account for induced velocity. It is likely that the angle of incidence (Vogel, 1994) was lower than our reported  $\alpha$ . While our calculations of  $C_L$  and  $C_D$  account for local induced velocity ( $\epsilon$ , Eqns 3 and 4), note that  $\alpha$  is reported without accounting for local induced velocity. Thus, for example, a calculation for our mid-upstroke propeller model using a Rankine–Froude momentum jet model (Usherwood and Ellington, 2002a; Vogel, 1994) suggests that local induced velocity at the wingtip would be  $\sim 1.26 \text{ m s}^{-1}$ . Therefore, the angle of incidence would be 52 deg, compared with 65 deg for  $\alpha$ . To explore the possible impact of induced velocity throughout the stroke cycle, we present a simulated  $\alpha$  throughout the stroke cycle (Fig. 3A, dotted line). This rough calculation using the same momentum jet model accounts for induced velocity during the downstroke, assuming that the induced velocity of the



downstroke heavily influences the upstroke as it returns through the downstroke wake. In general, these effects appear minimal ( $\sim 10$  deg), but have most pronounced effects at the downstroke–upstroke transition. Empirical measures of induced velocity would improve understanding of wing function *in vivo*, but will require methods such as PIV (Spedding and Hedenström, 2009) or hot-wire anemometry (Norberg et al., 1993).

### CONCLUSIONS

The propeller, while a simplification of the wingbeat, provides new insight into the functional significance of the tip-reversal upstroke in avian flight. It has often been assumed that the tip-reversal posture in birds is simply to reduce drag by allowing air to pass through individual feathers, creating a Venetian blind effect (Brown, 1953; Brown, 1963). Thus far, hummingbirds are the only birds that are known to use an aerodynamically active upstroke during slow flight (Warrick et al., 2005). Our new measurements indicated that tip-reversal upstroke, widespread in birds with pointed wings, might be more similar to the hummingbird wingbeat than previously thought. If an aerodynamically active upstroke does offer advantages, why is it that not all birds use a tip-reversal? A future challenge will be to better understand the trade-offs that have led to many species with rounded wings using a flexed-wing upstroke.

### LIST OF SYMBOLS

$c$	chord length
$C_D$	coefficient of drag
$C_h$	mean horizontal force coefficient
$C_L$	coefficient of lift
$C_v$	mean vertical force coefficient
$D$	drag
$F_h$	horizontal force
$F_R$	resultant force
$F_v$	vertical force
$L$	lift
$Q$	torque about the motor
$r$	radial distance along the wing
$R$	wing length
$S$	wing area
$S_2$	second moment of area
$S_3$	third moment of area
$V_T$	wing translational velocity
$\alpha$	geometric angle of attack
$\varepsilon$	downwash angle
$\rho$	air density
$\Omega$	angular velocity

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## RESEARCH ARTICLE

## Kinematics and aerodynamics of avian upstrokes during slow flight

Kristen E. Crandell\* and Bret W. Tobalske

## ABSTRACT

Slow flight is extremely energetically costly per unit time, yet highly important for takeoff and survival. However, at slow speeds it is presently thought that most birds do not produce beneficial aerodynamic forces during the entire wingbeat: instead they fold or flex their wings during upstroke, prompting the long-standing prediction that the upstroke produces trivial forces. There is increasing evidence that the upstroke contributes to force production, but the aerodynamic and kinematic mechanisms remain unknown. Here, we examined the wingbeat cycle of two species: the diamond dove (*Geopelia cuneata*) and zebra finch (*Taeniopygia guttata*), which exhibit different upstroke styles – a wingtip-reversal and flexed-wing upstroke, respectively. We used a combination of particle image velocimetry and near-wake streamline measures alongside detailed 3D kinematics. We show that during the middle of the wingtip-reversal upstroke, the hand-wing has a high angular velocity ( $15.3 \pm 0.8 \text{ deg ms}^{-1}$ ) and translational speed ( $8.4 \pm 0.6 \text{ m s}^{-1}$ ). The flexed-wing upstroke, in contrast, has low wingtip speed during mid-upstroke. Instead, later in the stroke cycle, during the transition from upstroke to downstroke, it exhibits higher angular velocities ( $45.5 \pm 13.8 \text{ deg ms}^{-1}$ ) and translational speeds ( $11.0 \pm 1.9 \text{ m s}^{-1}$ ). Aerodynamically, the wingtip-reversal upstroke imparts momentum to the wake, with entrained air shed backward (visible as circulation of  $14.4 \pm 0.09 \text{ m}^2 \text{ s}^{-1}$ ). In contrast, the flexed-wing upstroke imparts minimal momentum. Clap and peel in the dove enhances the time course for circulation production on the wings, and provides new evidence of convergent evolution on time-varying aerodynamic mechanisms during flapping in insects and birds.

**KEY WORDS:** Lift, Thrust, Clap and peel, Take-off, Wingtip-reversal, Flexed-wing, Particle image velocimetry

## INTRODUCTION

Takeoff, landing and slow flight are critical phases for all flying animals, whether foraging, evading predation or finding a mate. Slow flight ( $<3 \text{ m s}^{-1}$ ) is also an energetically costly form of flight (Tobalske et al., 2003). The aerodynamics of this important form of flight are not well understood because the majority of research on bird flight aerodynamics has focused either on hovering in hummingbirds (Warrick et al., 2005, 2009) or on medium- and high-speed flight (Spedding et al., 2003a,b; Spedding and Hedenstrom, 2009; Henningson et al., 2011; Spedding and Hedenstrom, 2009). Here, we set out to test the hypothesis that there is aerodynamic activity during the general avian (i.e. non-hummingbird) upstroke, an idea proposed first by Marey (1890) and subsequently by Lorenz (1933) and Brown (1953). We describe the kinematic and

aerodynamic patterns of slow flight in two species of birds, the diamond dove and the zebra finch, with two distinct upstroke styles: wingtip-reversal and flexed-wing upstroke.

When engaged in slow flight, birds generally do not maintain an extended wing. Instead, all birds except hummingbirds (Warrick et al., 2005) fold or flex their wings on the upstroke (Tobalske, 2007). During medium and fast flight in many species, the wing is kept partially extended during upstroke, and so is understood to be aerodynamically active (Spedding et al., 2003b), particularly at faster speeds (Henningson et al., 2011). In contrast, the observed kinematic patterns during slow flight have prompted the well-established prediction that the upstroke is aerodynamically inactive, halving the amount of time during a wingbeat that a bird can produce useful forces. Few direct measures of the kinematics and aerodynamics of upstroke in slow flight exist, however (Chang et al., 2013; Hedenström et al., 2006; Lentink et al., 2015; Muijres et al., 2012a,b,c; Spedding et al., 2003a). Most studies of takeoff flight focus on the immediate transition from legs to wings (Berg and Biewener, 2010; Earls, 2000; Provini et al., 2012).

During slow flight, most birds have one of two upstroke patterns: a flexed-wing upstroke or a wingtip-reversal upstroke. During the flexed-wing upstroke, the entire wing is pulled toward the body. Birds with this pattern generally have low aspect ratio, rounded wing tips (Brown, 1963). A study of slow flight in flycatchers (*Ficedula hypoleuca*) found no evidence for an aerodynamically active flexed-wing upstroke (Muijres et al., 2012a). The flexed-wing upstroke becomes aerodynamically active gradually as flight speed increases from  $4 \text{ m s}^{-1}$  to  $8 \text{ m s}^{-1}$  in both the thrush nightingale (Spedding et al., 2003b) and the robin (Hedenström et al., 2006), yet aerodynamic activity at speeds less than  $4 \text{ m s}^{-1}$  remains untested. The contrasting wingtip-reversal upstroke style keeps the distal wing extended, with the primary feathers supinated but traveling through an arc in the air. Birds with pointed, high aspect-ratio wingtips tend to exhibit a tip-reversal upstroke during slow flight (Simpson, 1983; Tobalske, 2007); galliformes are an exception (Tobalske and Dial, 2000).

Several lines of evidence suggest the tip-reversal upstroke produces aerodynamic force. Feather loading in live pigeons (*Columba livia*) and propeller models using pigeon wings suggest that the tip-reversal upstroke is aerodynamically active (Coming and Biewener, 1998; Crandell and Tobalske, 2011). *In vivo* accelerometers mounted to the trunk of a cockatiel (*Nymphicus hollandicus*) also indicate that the tip-reversal upstroke produces 14% of the net force of downstroke (Hedrick et al., 2004). Lastly, kinematic analyses of pigeons suggest the upstroke produces 50% of the net force of the downstroke during slow-speed maneuvering (Ros et al., 2011). However, the aerodynamic mechanism remains unknown.

It has been hypothesized, based on wing kinematics (Bennett, 1977; Scholey, 1983; Simpson, 1983), that birds which produce a tip-reversal upstroke capitalize on a time-varying mechanism called clap and fling (or clap and peel). This mechanism creates circulation at the transition from upstroke to downstroke via an interaction

Field Research Station at Fort Missoula, Division of Biological Sciences, University of Montana, Missoula, MT 59812, USA.

\*Author for correspondence (kristen.crandell@gmail.com)

**List of symbols and abbreviations**

$A$	area swept by the wing
$c$	wing chord
$F$	net force
PIV	particle image velocimetry
$T$	time interval
$u$	average airspeed
$V_A$	angular velocity
$V_T$	global wingtip speed
$\alpha$	global angle of attack
$\beta$	angle between wings
$\Gamma$	circulation
$\rho$	air density
$\omega$	vorticity

between the left and right wings. Similar insect kinematic patterns were first observed in *Drosophila melanogaster* and described by Vogel (1967). Weis-Fogh described this pattern in the chalcid wasp *Encarsia formosa* and predicted this motion increases the strength of circulation around the wing (Weis-Fogh, 1973). Subsequent models indicated the fling is aerodynamically active (Maxworthy, 1979) by initiating circulation immediately along the wing. This circulation contributes heavily to a leading-edge vortex that grows throughout the downstroke (Sane, 2003). Clap and fling is observed in a large insect, the hawkmoth ( $Re=8000$ ) (Bomphrey et al., 2006). Theory and empirical measurements with flapper models both indicate a boost in force production via this mechanism (Ellington, 1984; Lehmann et al., 2005). Among birds, there exists aerodynamic evidence that Japanese white-eye (*Zosterops japonicus*) and Gouldian finch (*Erythrura gouldiae*) capitalize on a ventral clap, wherein the wings are brought together beneath the body to contribute to the downwash (Chang et al., 2013). Here, we differentiate between a clap and fling versus a clap and peel, as defined by Miller and Peskin (2009): we define a clap and fling as minimal flexibility in a rigid wing structure, wherein changes in shape are from external mechanisms, such as aerodynamic loading and inertia; clap and peel, then, has flexibility spanwise and chordwise along the wing, and shape changes are both passive and active via musculoskeletal control.

We undertook the present study to test for aerodynamic activity in both upstroke styles, and describe the kinematics in relation to the forces they produce. We tested the following hypotheses: the kinematic clap and peel pattern yields flow regimes similar to the clap and peel of insect flight, despite dramatic differences in body size and, thus, Reynolds number (hawkmoth  $Re=8000$ , dove  $Re=51,000$ ). Additionally, the flexed-wing upstroke will be aerodynamically inactive. In both birds, we additionally expected the aerodynamic circulation (or lack thereof) produced during the upstroke would carry over and impact the circulation produced during the downstroke. To address these hypotheses, we compared two species: the diamond dove [*Geopelia cuneata* (Latham 1801), 40 g], which exhibits a wingtip-reversal upstroke, and the zebra finch (*Taeniopygia guttata* Reichenbach 1862, 13 g), which uses a flexed-wing upstroke.

**RESULTS****Kinematics**

We analyzed kinematic patterns for the third wingbeat in a flight sequence for four birds of both species (morphometrics available in Table 1). All flights were within  $\pm 5$  deg of level flight. In kinematic

**Table 1. Morphometrics for individual birds**

	Mass (g)	Wing area (cm <sup>2</sup> )	Wing length (cm)	Wing chord (cm)	Aspect ratio
Dove					
2	43.3	46.81	10.56	5.79	0.99
3	38.7	38.67	9.65	4.99	1.10
4	40.3	60.14	11.54	5.62	0.83
5	38.2	75.45	11.83	6.59	0.68
Mean $\pm$ s.d.	40.1 $\pm$ 2.3	55.3 $\pm$ 16.1	10.9 $\pm$ 1.0	5.7 $\pm$ 0.6	0.9 $\pm$ 0.2
Finch					
1	13	29.41	6.92	4.69	1.08
2	12.2	28.7	6.62	5.8	1.06
3	12.1	36.79	8.44	5.32	1.03
4	12.5	29.88	8.04	4.59	1.21
Mean $\pm$ s.d.	12.5 $\pm$ 0.4	7.5 $\pm$ 0.9	5.1 $\pm$ 0.6	5.1 $\pm$ 0.6	1.1 $\pm$ 0.1

All birds were used in particle image velocimetry (PIV) and kinematic analyses. Dove 4 and finch 1 were used in smoke-line flow studies.

trials, the doves had an average body velocity of  $1.54\pm 0.36$  m s<sup>-1</sup>, while the finches had an average body velocity of  $1.03\pm 0.28$  m s<sup>-1</sup>. During the third wingbeat, the doves were still accelerating at an average of  $5.5\pm 0.54$  m s<sup>-2</sup>, and finches were accelerating at  $4.34\pm 1.54$  m s<sup>-2</sup>.

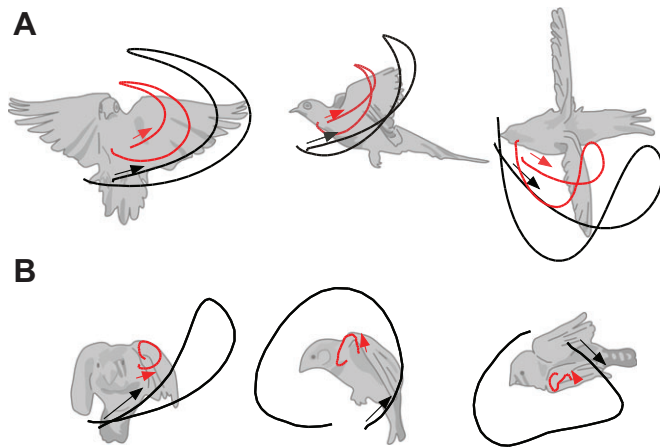
Three-dimensional traces of the path of the wingtips of the dove and finch followed the general pattern of the lateral-view kinematics of the rock pigeon and black-billed magpie (*Pica hudsonia*), respectively, as described in Tobalske and Dial (1996). Our 3D analysis revealed a figure-of-eight pattern in the dorsal and sagittal plane of the dove, but not in the finch (Fig. 1). The finch pattern produced a near-complete semi-circle in both the dorsal and sagittal planes, where the wingtip (black line) was kept caudal to the body during the upstroke, and oriented cranially during downstroke. In the dove, the wingtip transitioned from cranial to caudal during the upstroke (Fig. 1, dorsal and sagittal planes). A further difference between the species was visible in the transverse plane. In the dove, the hand-wing was kept extended ( $80.2\pm 3.2\%$  of wing length) during both upstroke and downstroke. In contrast, the finch flexed the wing ( $64.8\pm 5.2\%$  of wing length) during upstroke, most noticeably in the wrist kinematics (Fig. 1, transverse plane).

The global angle of attack ( $\alpha$ , defined as the angle between the velocity vector of the hand-wing and the wrist-secondary feather chord line) differed between species during upstroke, but followed a similar trend (Fig. 2A). The dove had a negative  $\alpha$  during upstroke, with a minimum of  $-81.3\pm 4$  deg during the first third of upstroke. The dove transitioned from a negative to a positive  $\alpha$  at wing turn around. In the finch, during upstroke,  $\alpha$  remained negative, with a minimum value of  $-45.9\pm 7.3$  deg. As the transition between upstroke and downstroke approached,  $\alpha$  transitioned through 0 deg, peaking during the downstroke at  $51.5\pm 7.4$  deg.

Global wingtip speed ( $V_T$ ) showed similar patterns between the dove and finch (Fig. 2B). In the dove, there were two peaks in  $V_T$ , the first during upstroke at  $8.4\pm 0.6$  m s<sup>-1</sup>, and the second during downstroke at  $9.9\pm 1.3$  m s<sup>-1</sup>. These indicate maximal  $V_T$  produced during the middle of the limb swing phase, at the transition from limb acceleration to limb deceleration. The finch also exhibited two peaks, although the peak during upstroke remained for approximately 5 ms. During the upstroke, peak  $V_T$  was  $7.5\pm 3.3$  m s<sup>-1</sup>. During the downstroke,  $V_T$  peaked at  $11.0\pm 1.9$  m s<sup>-1</sup>. This peak occurred following the transition from upstroke to downstroke, and so took place relatively earlier in the finch than the dove.

Angular velocities ( $V_A$ ) were defined as the angular change between the orientation of the position vector of the hand-wing





**Fig. 1. Traces of representative bird-centered wingtip and wrist movements of the study species.** (A) Diamond dove, (B) zebra finch. Traces (left to right) are from the transverse, sagittal and dorsal planes (bird silhouettes represent the mid-upstroke position, and are not an exact representation of posture at that point in time). Wrist is represented in red, wingtips in black. Arrows indicate the start of upstroke (transition from downstroke, time 0%).

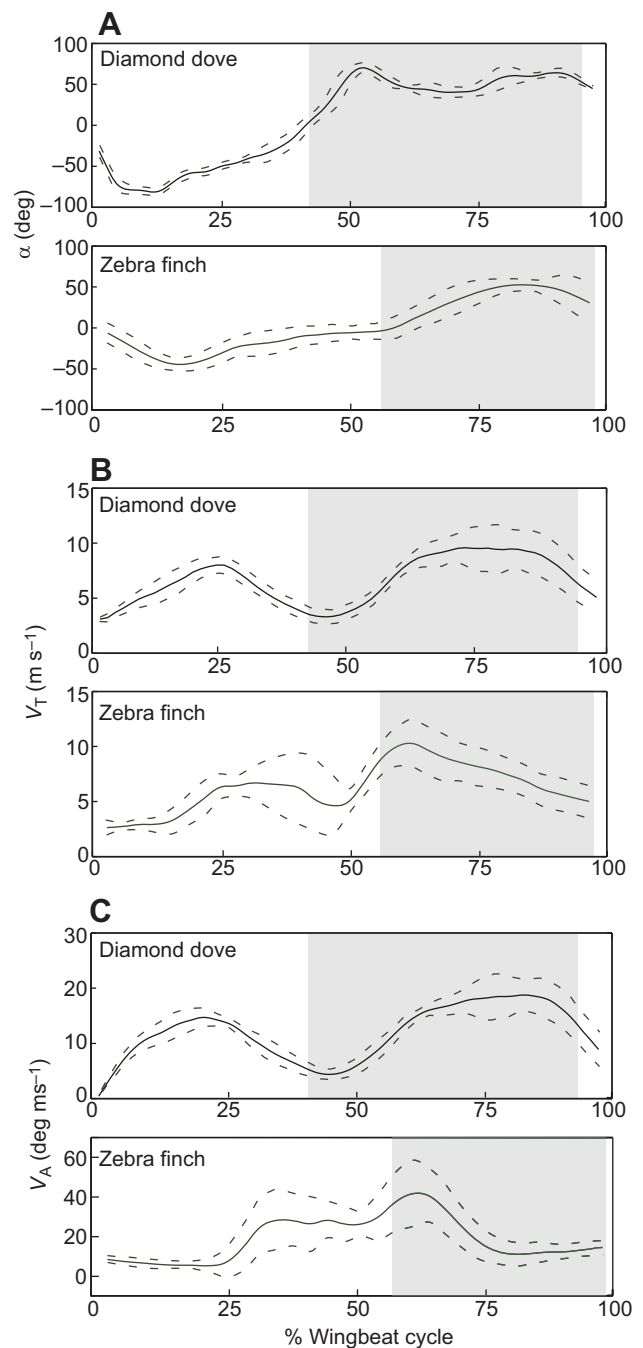
centroid points, in global coordinates. These velocities followed a similar pattern to wingtip speed in the dove, but not the finch (Fig. 2C). In the dove, two peaks in  $V_A$  occurred, in the same pattern as  $V_T$  – peaking halfway through upstroke at  $15.3 \pm 0.8 \text{ deg ms}^{-1}$  and halfway through downstroke at  $19.4 \pm 2.1 \text{ deg ms}^{-1}$ . In the finch, however, only one prominent peak in  $V_A$  occurred, at  $45.4 \pm 13.8 \text{ deg ms}^{-1}$ , at the transition between upstroke and downstroke. At this point, the wing rapidly reoriented from being strongly adducted to being fully extended. A minor  $V_A$  peak occurred during the upstroke at  $32.2 \pm 9.7 \text{ deg ms}^{-1}$ , but this peak remained relatively stable until the transition between upstroke and downstroke.

Our analysis of wing–wing interactions in three dimensions illustrates each portion of the clap and peel cycle in the dove, relative to wing contact time (Fig. 3). In the wingtip-reversal at the end of the upstroke, the rapidly extending wings came into contact (the ‘clap’), with the left and right wing leading edges coming together approximately 3 ms before the trailing edge, at approximately 5% of the entire wingbeat cycle. The wingtip leading edges were  $1.5 \pm 0.6 \text{ cm}$  apart (approximately the width between the shoulders) at closest contact, while the trailing edges reached almost full contact at  $0.2 \pm 0.1 \text{ cm}$  apart. Fig. 3A illustrates the leading edge (red, orange) on the right and left wing coming into contact (red arrow). The trailing edge (Fig. 3A, blue, green) came together later (blue arrow). Following the clap, the wings moved apart during the peel phase as downstroke began. The wings first separated at the leading edge (Fig. 3A, red arrow), with the trailing edge separation occurring 4–9 ms later (blue and green arrows).

In contrast, during the flexed-wing upstroke of the zebra finch, all points along the wing stayed close to the body during upstroke, with the wingtip at  $3.2 \pm 0.4 \text{ cm}$  between the left and right wings (approximately the width of the body, Fig. 3B). At the transition from upstroke to downstroke, the wings opened approximately  $6.2 \pm 0.7 \text{ cm}$  apart. Notably, the trailing edges of the finch did come close to contact during the transition from upstroke to downstroke (Fig. 3B, blue and green).

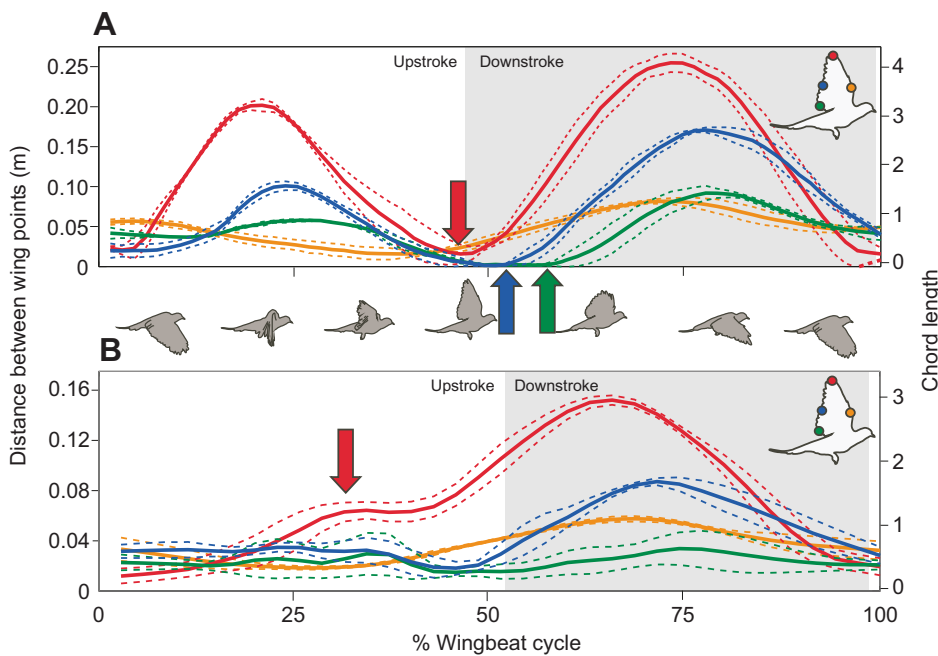
### Aerodynamics

We analyzed the wake of four birds for each species (morphometrics available in Table 1). Zebra finches averaged a body velocity of



**Fig. 2. Kinematic variables for the centroid of the hand-wing.** (A) Angle of attack ( $\alpha$ ), (B) translational speed (wingtip speed,  $V_T$ ) and (C) angular velocity ( $V_A$ ). Data were extracted from the diamond dove and zebra finch during the third wingbeat of the downstroke. Solid lines represent means, dashed lines represent the standard deviations. The shaded area represents the downstroke.

$1.03 \pm 0.28 \text{ m s}^{-1}$  and diamond doves averaged a body velocity of  $1.54 \pm 0.36 \text{ m s}^{-1}$ ; both species were accelerating. Flights analyzed with smoke visualization maintained free-stream velocities of approximately  $0.95 \pm 0.05 \text{ m s}^{-1}$ . All flights analyzed were within  $\pm 5 \text{ deg}$  of level flight. We observed significant aerodynamic activity in the wingtip-reversal upstroke of the diamond dove, and minimal activity during the flexed-wing upstroke of the zebra finch. In the dove, a shed vortex was evident in our particle image velocimetry (PIV) images (Fig. 4) at the transition from upstroke to downstroke (circulation  $\Gamma = 14.4 \pm 0.09 \text{ m s}^{-2}$ ). This same type of



**Fig. 3. Distances between opposite wings during the third full wingbeat cycle in diamond doves and zebra finches.**

(A) Distance between wing points in the diamond dove. Note the significant lag time between leading edge separation (red, orange) and trailing edge separation (blue, green). Arrows show contact and separation of wingtips (red) before trailing edges (blue, green). (B) Distance between wing points in the zebra finch. Note the distinct kinematic differences from the dove during upstroke, as the finch holds its wings close to the body during this phase (orange, blue, green). The leading edge wingtip (red) is held close to the body during the start of upstroke in the finch, and these edges begin to separate over 2/3 of the way through the upstroke time (red arrow). Means and s.e. are represented by solid and dashed lines, respectively. The scale on the right is relative to the wing chord (Table 1). Bird outlines between graphs are modeled after the diamond dove.

upstroke-generated vortex was never observed in PIV images from the zebra finch (Fig. 5).

During the last half of upstroke, the diamond dove hand-wing was aerodynamically active in that it displaced streamlines (Fig. 6A). Induced velocity to the air was visible as an abrupt cut-off between the smoke lines through which the wing moved, visible at 35% and 45% of the wingbeat cycle in Fig. 6A (red arrowhead and ellipse). At the end of upstroke (Fig. 6A, 55%), the wings came into contact, and the entrained air around the wings was shed. As described above, we detected this shed vortex in our parasagittal samples of PIV images as circulation deposited into the wake at the transition between upstroke and downstroke in the diamond dove (Fig. 4A). Induced velocity behind the wing was also visible, at  $2.73 \pm 0.23 \text{ ms}^{-1}$  for an arbitrarily selected patch immediately behind the wings (Fig. 7).

From smoke lines near the finch, it was evident that the feathers act similar to those in the doves, imparting momentum to the wake; smoke was entrained on the primary feathers during upstroke (Fig. 6B, 10 ms, red arrowhead; supplementary material Movie 1). However, the surface area of the wing was much reduced in comparison to the dove (Fig. 6).

Both species created a starting vortex during the first part of downstroke (Fig. 4B and Fig. 5B), but the initiation of circulation appeared to be relatively earlier in the dove (Fig. 4A), which we interpret as being due to the clap and peel pattern of the wings, approximately concurrent with the shedding of circulation into the wake. We first measured this circulation in a parasagittal plane over the back of the bird as  $26 \pm 3\%$  the value of the final downstroke circulation (Fig. 8). In the dove, this circulation grew throughout the downstroke (Fig. 4), and reached values that were not significantly different from the final shed vortex halfway through the downstroke. In contrast, circulation measured in the zebra finch in the same plane was approximately 0% of final circulation values at the start of early downstroke (Fig. 5A and Fig. 8). This was a consequence of the lack of an initial shed vortex created during upstroke. However, as observed in the smoke line video, the finch had a start vortex that began independently on both wings, just like the dove (supplementary material Movie 1). As a consequence of this right and left wing-

independent vortex ring structure (Poumazeri et al., 2013), the measured circulation in the sagittal plane dorsal to the bird was 20–40% lower in the finch at the start of downstroke, suggesting a relatively smaller time interval over which circulation was created. This smaller time frame was a consequence of the wings of the finch beginning the downstroke farther from the midline of the bird (Fig. 3).

#### Force estimates in the dove

To provide a rough estimate of the net forces that the clap and peel mechanism may provide relative to the subsequent downstroke, we adopted two models: a momentum-jet model (Vogel, 1994) focusing on downstroke, and a reverse peel model (Miller and Peskin, 2009), focusing on the clap. Both models assume steady-state flow conditions and circulation averages are representative of the wingbeat. As such, they are simplifications, and caution is necessary in interpreting the results; our goal was to provide initial quantitative estimates to help motivate future research that may employ time-resolved and volumetric techniques for flow measurement (e.g. Hubel et al., 2010; Bomphrey et al., 2012).

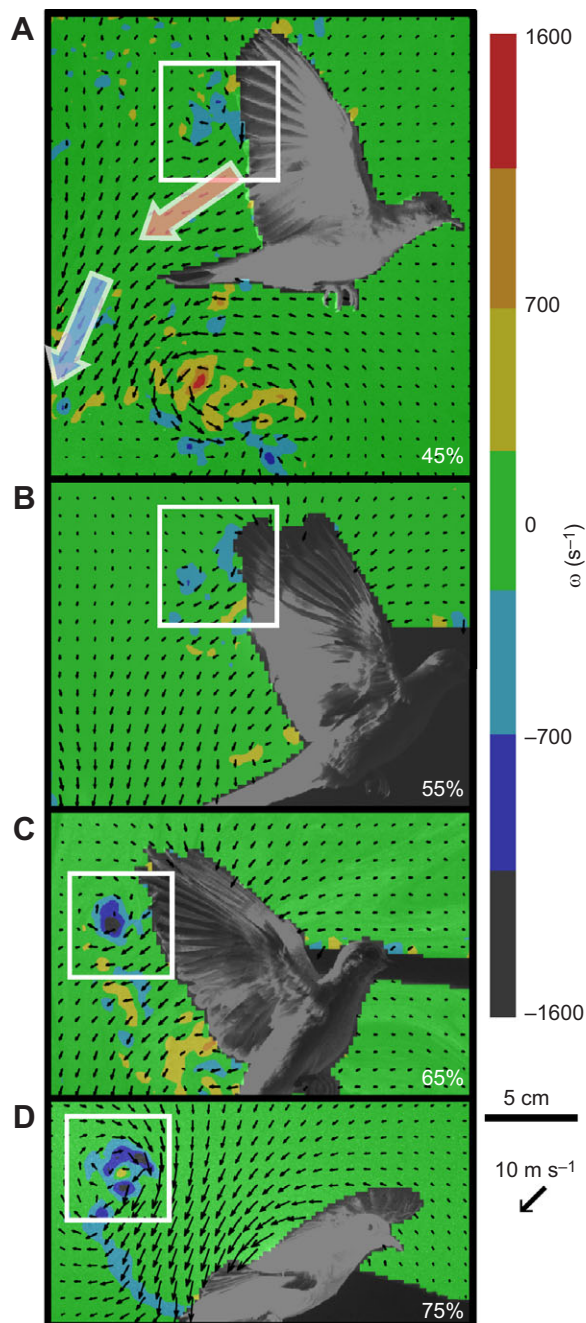
We estimate average force produced during the downstroke phase to be  $0.74 \text{ N}$  – 188% body weight for the average dove weight of  $0.39 \text{ N}$ . This is consistent with body weight estimates for the downstroke in a dove during steady flight on a level path given that downstroke is 60% of the total wingbeat cycle (Provini et al., 2012).

For the contribution of the clap in the reverse peel model, we incorporated the angle between the wings (50 deg) and induced velocity ( $2.5 \text{ m s}^{-1}$ ; Fig. 7). With these values, the reverse peel model indicates the clap phase may produce  $0.35 \text{ N}$ , or 89% of body weight. The reverse peel model does not include an estimate of force production during the translational phase of the upstroke, and so is likely an underestimate. We estimate the time duration for this average force production to be approximately 10 ms, or 15% of the total wingbeat cycle. In other words, the impulse from downstroke is approximately  $0.031 \text{ N}$ , while the impulse from upstroke is  $0.0035 \text{ N}$ .

#### DISCUSSION

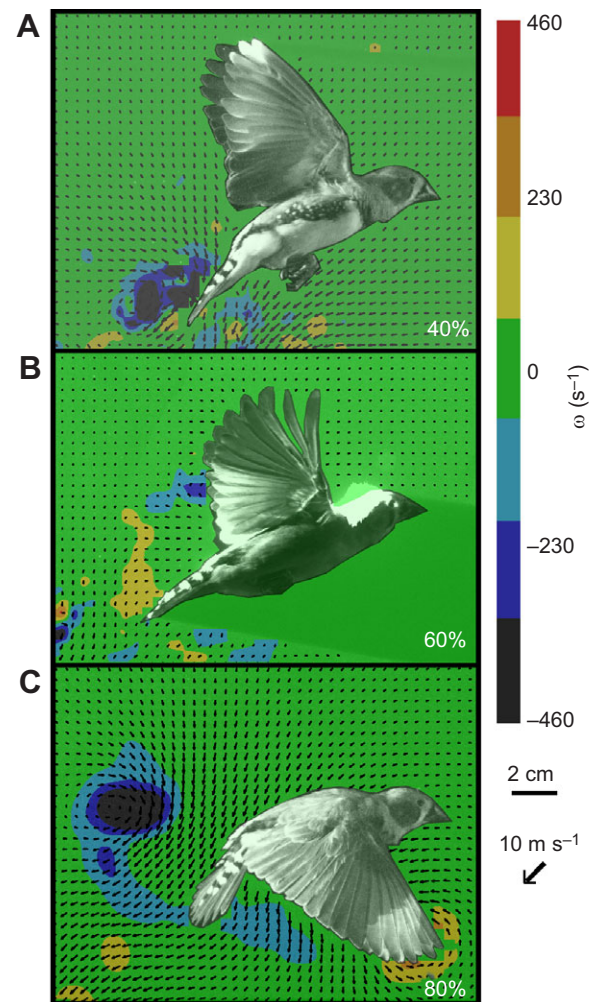
In the two bird species we studied, we observed two distinct aerodynamic patterns, consistent with distinct wing kinematics.





**Fig. 4. A sequence of wake visualizations for the diamond dove clap and peel stroke cycle transition.** The sequence was assembled using data from two individuals. Vectors in the foreground represent air velocity ( $\text{m s}^{-1}$ ) and colors in the background indicate vorticity ( $\omega$ ,  $\text{s}^{-1}$ ). White boxes indicate the vortex shed during the clap (A) and the downstroke vortex growing during the subsequent peel (B,C) and downstroke (D). Large arrows in A summarize the direction of airflow (blue, downwash from a previous downstroke; red, velocities imparted by the upstroke). The approximate percentage of the wingbeat cycle from kinematics in Figs 2 and 3 is indicated.

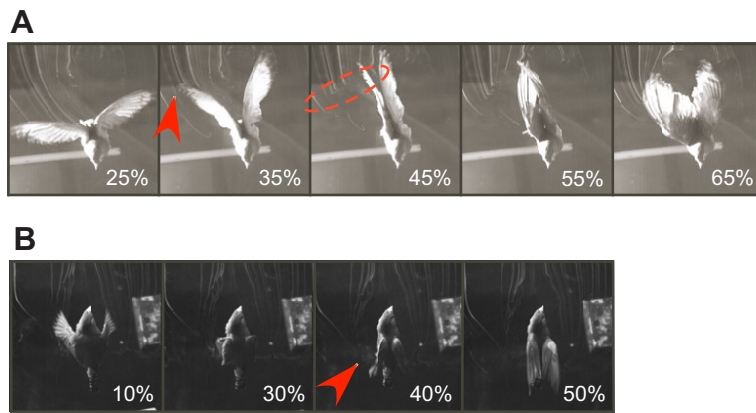
One, the wingtip-reversal upstroke in the dove, produces measurable circulation, evident as a vortex core and induced velocity approximately concurrent with the animal clapping and peeling its wings apart (Figs 4, 7). In contrast, some aerodynamic activity was apparent in streamlines during the flexed-wing upstroke of the zebra finch (Fig. 6B), but this activity was so minimal as to not be manifest in our parasagittal PIV images of the wake from this



**Fig. 5. A sequence of wake visualizations for the zebra finch upstroke to downstroke transition.** The sequence was assembled using data from one individual. Vectors in the foreground represent air velocity ( $\text{m s}^{-1}$ ) and colors in the background indicate vorticity ( $\omega$ ,  $\text{s}^{-1}$ ). Note a lack of vorticity at the transition from upstroke to downstroke (A), and a lack of initiation of circulation during the first quarter of downstroke (B), until final levels are reached at the end of downstroke (C). The approximate percentage of the wingbeat cycle from kinematics in Figs 2 and 3 is indicated.

species (Fig. 5). We found support for our initial hypothesis that the clap and fling style kinematic pattern produces potentially useful aerodynamic forces (estimated at an average of 89% body weight produced during the clap). Further, we found that this capacity to generate aerodynamic force in late upstroke then continues uninterrupted throughout the subsequent downstroke. We found support for our second hypothesis in that the flexed-wing upstroke of the finch produces negligible aerodynamic forces, and as a consequence leads to relatively delayed growth in circulation during the downstroke.

We present  $\alpha$  and  $V_T$  data for the finch (Fig. 2), but the negligible aerodynamic forces observed suggest that these measures may be irrelevant during the majority of the upstroke, particularly as the wing is kept near the body and so is likely not extended enough to induce momentum to the wake. Although evidence was lacking in our lateral PIV measures, the finch downstroke may be aerodynamically representing a distant-peel condition, wherein the left and right wing do not directly contact, but may be close enough to manifest an aerodynamic benefit similar to the peel



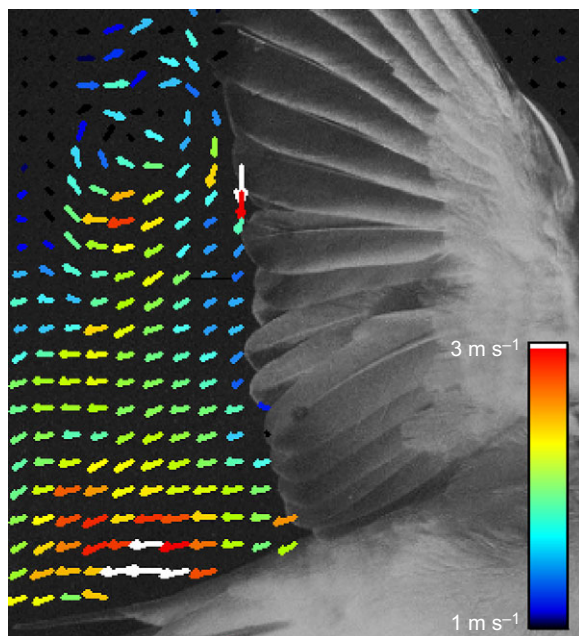
**Fig. 6. Smoke-line traces of airflow during flight.** (A) In the diamond dove, induced velocity is evident mid-way through the upstroke as a deflection of smoke lines (arrowhead) at 35%, and the disrupted wake visible at 45% (ellipse). (B) In the zebra finch, minimal aerodynamic activity is seen during the upstroke. A small amount of aerodynamic activity is visible at 40% (arrowhead). Approximate percentages of the wingbeat cycle are noted in conjunction with previous figures.

condition (Scholey, 1983). In particular, the peak wingtip velocity in the finch occurs during this phase, and may in fact contribute to the initial creation of circulation. Two passerine species demonstrate clap kinematics at the transition from downstroke to upstroke (Chang et al., 2013), so it is feasible that a similar pattern occurs during the upstroke to downstroke transition. Further work with high-speed, stereo PIV is necessary to explore this hypothesis. Wing inertia may play a crucial role in wing–wing contact (Hedrick et al., 2004); this may serve as an explanation for why short, rounded wing species use a flexed-wing upstroke, and long, pointed wing species typically use a wingtip-reversal upstroke (with the notable exception of the galliformes; Tobalske and Dial, 2000).

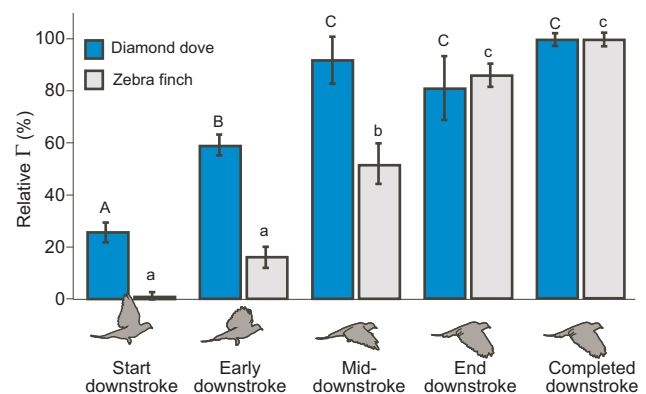
The kinematic patterns of the clap and peel during the transition from upstroke to downstroke appear different between insects and birds, likely because of different musculoskeletal arrangements. Birds actively pronate the leading edge of the wing to begin the peel phase, with the trailing edge following. Muscles in the distal wing are active during this phase (Dial, 1992), suggesting muscular control of the distal wing. Insects, however, do not have controlling muscles in their wings, and must rely on forces from indirect

muscles deforming the trunk or direct muscles acting at the wing base to separate the two wings (Dudley, 2000). This morphological dissimilarity presents a difference in kinematics. For example, in *E. formosa* during transition from upstroke to downstroke, the clap phase alone constitutes between 20% and 25% of the entire wingbeat cycle (Ellington, 1975). In the dove, this phase is only 2% of the total wingbeat cycle. Further, unlike in theoretical models (Ellington, 1984), the vorticity produced by the clap phase in the dove appeared to be a direct result of shedding circulation created during the upstroke. This is visible in Fig. 6 at 25–45%, and the wake left behind is seen in Fig. 4. In insects, the clap phase expels air from between the left and right wing, similar to a jet (Miller and Peskin, 2009). However, in the dove, circulation created during the upstroke is shed during the clap phase.

The functional contribution of tip-reversal in slow forward flight appears to be as thrust rather than weight support. As the wings are moving caudally, induced velocity in the wake is convected caudally and laterally (supplementary material Movies 2,3). The observed shed vortex is oriented behind the bird, evident in the PIV image (Fig. 4A, red arrow). Shedding of vorticity was likely assisted by the rapid motion of the two wings coming into contact dorsally (Fig. 3, red arrow). In sum, these wing motions and the resulting induction of velocity into the air would likely be beneficial to a bird seeking to move forward. This observation is corroborated by our



**Fig. 7. Induced velocities behind the wing of the diamond dove, caused by motion during upstroke.** Velocities behind the wing were created during the upstroke. The scale represents  $1\text{--}3\text{ m s}^{-1}$ .



**Fig. 8. Circulation during the wingbeat cycle in diamond doves and zebra finches.** Circulation ( $\Gamma$ ,  $\text{m}^2\text{ s}^{-1}$ ) was standardized by dividing by the value of completed downstrokes previously left in the wake. Clap and peel increases circulation starting at the end of upstroke (the 'start downstroke' phase) in the dove, but not the finch. Data are means $\pm$ s.e. Different letters represent statistically significant differences in mean circulation ( $P<0.05$ ) within species (uppercase, doves; lowercase, finches). Circulation for each phase was statistically different between species ( $P<0.01$ ) for all but the end of downstroke phase ( $P=0.94$ ) and completed downstroke ( $P=0.97$ ).



previous model of the tip-reversal upstroke (Crandell and Tobalske, 2011) and kinematic measurements of whole-body acceleration that are consistent with an aerodynamically active upstroke in pigeons, a species that uses tip-reversal (Ros et al., 2011). Recent measurements in parrotlets (*Forpus coelestis*), a species that exhibits tip-reversal in slow flight, indicates minimal weight support; thrust was not measured (Lentink et al., 2015). This mechanism has the potential to contribute to decelerations during braking, perhaps via interactions with the wake of the previous downstroke. It may additionally contribute to weight support during hovering, although slight kinematic differences likely persist to re-orient flow. The generalities of this mechanism thus merit further study.

Through circulation and vorticity shed via wing–wing interactions, we interpret the clap and peel mechanism supplements circulation produced during downstroke. If this is the case, additional aerodynamic activity may supply additional thrust, potentially providing greater control during slow flight maneuverability (Ros et al., 2011). Further, clap and peel may generate a larger active area swept by the wings, and hence a greater whole-wingbeat span efficiency (Henningsson and Bomphrey, 2011, 2013; Henningsson et al., 2014; Muijres et al., 2011). However, before we can fully understand the details of time-varying force production and span efficiency, it will be essential to obtain better-resolved flow data. Our study will hopefully motivate new tests including time resolution and sampling either the Trefftz plane (Hedenström et al., 2007) or, alternatively, measurements directly on the wing (Muijres et al., 2008; Warrick et al., 2009). Such techniques would offer the opportunity to better estimate instantaneous forces. Further research of the wake flow is necessary using high-speed systems (Hedenström et al., 2009; Henningsson et al., 2011; Hubel et al., 2010) and whole-volume sampling (Bomphrey et al., 2012) to more fully elucidate the three-dimensional and time-varying nature of force production.

Our results add a new example of evolutionary convergence with insects and bats on the use of aerodynamic mechanisms. The clap and peel pattern resulting in a vortex shed in the sagittal plane and induced velocity directed posterior to the animal is consistent with the time-averaged aerodynamic flow pattern created by clap and fling in hawkmoths (Bomphrey et al., 2006). Thus, clap and peel joins leading edge vortices generated during the translational phase of wing movement and rotational circulation during wing reversal (Warrick et al., 2009) as examples of convergence with insects. Bats use a similar wingtip-reversal kinematic pattern during slow flight; however, the aerodynamic activity that has been observed occurs during the translational phase of the hand-wing, producing induced flow behind the animal similar to the flow measurements we have observed before the clap and peel in the dove (Hedenström et al., 2007). As such, bats have likely converged on a similar aerodynamic pattern during upstroke – the generality of the clap and peel in bats, however, remains untested.

Time-varying aerodynamic effects, most notably the leading edge vortex, have been demonstrated in gliding swifts (Videler et al., 2004), a slow flying flycatcher (Muijres et al., 2012a), and modeled in a large, flapping goose (Hubel and Tropea, 2010). Our study extends the understanding of the generality of such convergence across body sizes (and Reynolds numbers), meaning that time-varying aerodynamics are not limited to the smallest of vertebrate fliers (Muijres et al., 2008; Tobalske et al., 2009; Warrick et al., 2009). Diamond doves are 10× the mass of hummingbirds and nectarivorous bats.

The functional contribution of tip-reversal upstroke relative to flexed-wing upstroke begs the question of why it is that not all birds use the tip-reversal and clap and peel mechanisms. Addressing this question will require new insight into the various selective pressures that maintain rounded, low aspect ratio wings in birds. For example, rounded wings are thought to be more useful for avoiding wing damage in cluttered environments. In this context, it is noteworthy that Galliformes with rounded wings use tip-reversal when they are engaged in explosive escape flight after take-off (Tobalske and Dial, 2000). Another explanation was proposed by Vazquez (1992), who suggests there may be a trade-off in function between wrist-bone morphologies allowing a locked-in wing during glide versus those that facilitate the tip-reversal upstroke. In general, though, doves and pigeons glide when engaged in intermittent flight, whereas finches and related small passerines tend to flex their wings and bound (Tobalske, 2001). Further work examining the skeletal function using techniques such as XROMM (Brainerd et al., 2010) is necessary to test the ‘wrist-lock’ hypothesis (Vazquez, 1992).

## Conclusions

It has long been recognized that bird species exhibit one of two general upstroke styles during slow flight (Brown, 1948), and here we have demonstrated detailed 3D kinematics and the aerodynamic consequences of these upstroke styles. The zebra finch uses a flexed-wing upstroke, and this upstroke period appears to produce a minimal aerodynamic trace. Diamond doves, with a wingtip-reversal upstroke, exhibit a clap and peel during the upstroke–downstroke transition that generates a shed vortex with measurable induced velocity added to the wake in a way that suggests thrust production that should help the bird progress on a forward path. Further work using time-resolved and volumetric PIV is necessary to refine this initial understanding of the aerodynamic contribution of the upstroke during slow-speed flight in birds.

## MATERIALS AND METHODS

### Animals and experimental design

Four diamond doves (*G. cuneata*,  $Re=51,000$ ) and four zebra finches (*T. guttata*,  $Re=8500$ ) were trained to fly between two perches positioned 2 m apart within a netted chamber (2×2×2 m) for all but the streamline experiment. For qualifying streamlines, birds flew in the working chamber of a variable-speed wind tunnel (0.6×0.6×0.8 m; see Tobalske et al., 1999, for details of the tunnel design). Information for the morphometrics of each bird is provided in Table 1; Reynolds number was calculated with mean flight velocity during kinematic trials and chord length is reported in Table 1. All procedures were approved by the Institutional Animal Care and Use Committee at the University of Montana.

### Kinematics

We recorded wing and body motion using four high-speed video cameras – a Photron SA-3, two PCI 1024 s and a Phantom Miro Ex4, all synchronized using a TTL pulse. Recordings were made at 1000 Hz with a shutter speed of 1/10,000 s. Kinematic analyses of the wings and body during the third wingbeat following take-off were reconstructed in Matlab using a DLT conversion (Hedrick, 2008).

To quantify wing motion, we extracted the global angle of attack ( $\alpha$ ), global translational wingtip ( $V_T$ ) and angular velocity ( $V_A$ ) of the wing in global coordinates for a centroid of the hand-wing in both species (supplementary material Fig. S1). Additionally, to facilitate comparison between species, the wing cycle was divided into relative segments, with 0% and 100% defined as the start of upstroke based on wrist turn-around. The transition between upstroke and downstroke was defined visually at the directional reversal of the wing, and occurred at approximately 28 ms in the dove and 17 ms in the finch. For simplicity, and because we could not obtain time-resolved, whole-volume measures of flow velocity, we did not include induced velocity in our measures of  $\alpha$  and  $V_T$ .

Several kinematic parameters were calculated for the hand-wing. To compare wing–wing contact, we digitized the wrist, wingtip (10th primary in the dove, 9th primary in the finch), first secondary feather and last secondary feather for both the left and right wing. To compare distances between different points on the wing, we calculated the linear distance between each contralateral point on the corresponding left and right wing.

For additional kinematic parameters, we utilized an average location of the hand-wing. We calculated a hand-wing centroid by averaging the  $x$ ,  $y$  and  $z$  position of the wrist, 10th primary (dove) or 9th primary (finch), and the 3rd primary on the left wing. For all parameters presented, we used global coordinates (i.e. not bird-centered). Supplementary material Fig. S1 illustrates each variable described herein.  $\alpha$  was calculated by measuring the angle between the hand-wing centroid's velocity vector and a wing chord approximated as a line between the wrist and first secondary feather (approximating a flat wing).  $V_T$  ( $\text{m s}^{-1}$ ) was calculated with the direct linear distance between the centroid from frame to frame per unit time (1 ms apart).  $V_A$ , or rotational velocity, was calculated as the angle between the hand-wing centroid position vector relative to the shoulder in subsequent frames (1 ms). For all parameters, means and standard deviations across individuals were calculated.

### PIV

We used a PIV system (LaVision GmbH, Goettingen, Germany) with DaVis 7.1 software. A dual-cavity pulsed 50 mJ Nd:YAG laser was used to illuminate a 2 mm thick flow field, with planar dimensions spanning a field of approximately  $35 \times 25$  mm. The laser was oriented behind the flight trajectory, leaving the animal's head in a shadow, so as not to damage its eyes. We seeded the air with particles of olive oil less than  $1 \mu\text{m}$  in diameter, generated at a rate of  $7 \times 10^{10}$  particles  $\text{s}^{-1}$  using a vaporizer fitted with a Laskin nozzle. Particle illumination was recorded using a  $1376 \times 1040$  pixel charged-coupled device (CCD) camera placed perpendicular to the illumination field. To calculate particle velocity, we used cross-correlation of paired images with an elapsed time between images of 500  $\mu\text{s}$ . We used a multi-pass filter with an initial interrogation area of  $64 \times 64$  pixels and a final area of  $16 \times 16$  pixels with 50% overlap.

We calculated velocity ( $\text{m s}^{-1}$ ), vorticity ( $\omega$ ,  $\text{s}^{-1}$ ) and circulation ( $\Gamma$ ,  $\text{m}^2 \text{s}^{-1}$ ) as described in Warrick et al. (2009) and Provini et al. (2014). Briefly, we integrated all same-sign  $\omega$  in a given PIV field within 1.5 chord lengths of peak  $\omega$  to measure  $\Gamma$ . We considered each negatively signed vortex core deposited in the wake during downstroke to represent the cross-section of an elliptical vortex ring shed from the trailing edge from the wing. In the case of  $\Gamma$  deposited in the wake due to upstroke, we used a visual cut-off of streamlines to quantify the presence of  $\Gamma$  (see fig. 3 in Provini et al., 2012). This anomalous vorticity was considered to be created from the motion of the upstroke during the wing-tip reversal upstroke and is likely the result of vorticity shed from the wingtip. Average free-stream background vorticity was measured at  $0.006 \pm 0.008 \text{ s}^{-1}$ , 1.5% of average peak vorticity measured in the wake for the birds. Thus, we applied no masking. Added mass of the vortex wake was not calculated, as we lacked adequate time resolution to measure displacement of shed vortices (Dabiri, 2005). It has been previously shown in chukar partridge (*Alectoris chukar*) that birds generate negligible aerodynamic added mass (Tobalske and Dial, 2007).

We sampled the near-wake of the birds in a parasagittal plane ( $\pm 1$  cm from sagittal plane). Samples were taken at 5 Hz, and were recorded during the initial 3rd to 5th wingbeat after takeoff from a perch, at body velocities of approximately  $3 \text{ m s}^{-1}$  for the doves and  $1.5 \text{ m s}^{-1}$  for the finches. To quantify growth in  $\Gamma$  with respect to time, we divided the wingbeat into four phases: (1) transition from upstroke to downstroke (dove  $N=19$ , finch  $N=9$ ; approximately 40–50% of the entire wingbeat cycle; Figs 2, 3); (2) first quarter of downstroke (dove  $N=30$ , finch  $N=16$ ; approximately 50–63% of the wingbeat cycle); (3) second quarter of downstroke (dove  $N=20$ , finch  $N=13$ ; approximately 63–75% of the wingbeat cycle); and (4) third and fourth quarters of downstroke (dove  $N=14$ , finch  $N=21$ ; approximately 75–100% of the wingbeat cycle) based on wingtip elevation and depression relative to the bird. We measured previously shed downstroke  $\Gamma$  left in the

wake for comparison with the most recently completed wingbeat (dove  $N=124$ , finch  $N=82$ ). We tested for differences in the time course of circulation between birds using a one-way ANOVA, with individual as a factor, and found no differences in mean downstroke force production among individuals (for doves:  $0.76 < P < 0.99$ , for finches:  $0.33 < P < 0.98$ ); thus, all birds were combined for final analyses. Herein, we report means  $\pm$  s.d. To compare phases within species, we used a one-way ANOVA with phase as a factor, followed by a Tukey *post hoc* test to determine significant differences between phases. To compare the percentage circulation during each phase between species, we used a Student's *t*-test.

### Time-resolved flow

To more fully observe flow characteristics with respect to time and volume, we employed a classic smoke streamline study (Barlow et al., 1999; Ellington et al., 1996). We installed a nichrome wire (60 gauge) horizontally across the front of the working section of a variable speed wind tunnel (Tobalske et al., 1999). This wire was connected to a 15 V power supply. We seeded the wire with mineral oil droplets approximately 0.5 cm apart. When current was applied to the wire, it produced heat, and the droplets burnt to create streamlines. For these experiments, air velocity in the tunnel was  $0.95 \pm 0.05 \text{ m s}^{-1}$ . One dove and one finch were flown within the tunnel, and the flight and smoke flow were recorded from a dorsal view using a Photron PCI 1024 high-speed camera recording at 1000 frames  $\text{s}^{-1}$ , with a shutter speed of 1/5000 s.

### Force estimates

To gain an approximation of body weight support provided by the aerodynamic mechanisms we observed, we utilized two models. To estimate net force during the downstroke, we used a momentum jet model, as described by Vogel (1994):

$$F = \rho \Gamma A / T. \quad (1)$$

We used this calculation for each quarter of the downstroke divided equally by time ( $T=9.25$  ms), with corresponding measures of  $\Gamma$  (see Fig. 8) and swept wing area  $A$ , for each quarter, where air density  $\rho=1.07 \text{ kg m}^{-3}$  in Missoula, MT, USA. We then averaged the momentum jet model calculated for each section, for a net force estimate.

As a rough estimate of potential force production of the clap phase, we elected to use the reverse peel model (Miller and Peskin, 2009). This estimation was only possible for the dove, as the finch did not have a kinematic pattern close to a clap and peel, and so we could not estimate the angle between the two wings. The reverse peel is modeled as:

$$F = \rho \beta c u^2, \quad (2)$$

where  $\beta$  is the estimated angle between the two wings for the last 10 ms of the clap (approximately 50 deg in the dove),  $c$  is the wing chord and  $u$  is the average airspeed resulting from the clap (Fig. 7).

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### Competing interests

The authors declare no competing or financial interests.

### Author contributions

Both authors were involved in the research design, interpretation of findings and preparation of the manuscript. K.E.C. additionally performed the experiments and analyzed the data.

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## Supplementary material

Supplementary material available online at  
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1 **Skeletal kinematics of the avian wing**  
2 **during slow flight**  
3

4  
5 **Abstract**

6 During slow flight, birds exhibit predictable downstroke kinematics, but have  
7 one of two upstroke styles. These styles contribute to differences in aerodynamic  
8 force production, and so overall variation flight performance. The functional  
9 difference between upstrokes has been explained with variation in underlying  
10 skeletal morphology. Birds with an aerodynamically active “wingtip-reversal”  
11 upstroke style tend to exhibit greater distal muscle mass and more robust wing  
12 skeletal elements, in comparison to birds with the aerodynamically inactive “flexed-  
13 wing” upstroke style. Here, we address the hypothesis that the underlying skeletal  
14 morphology reflects differences in internal skeletal kinematics in two species: the  
15 Rock dove (wing-tip reversal upstroke) and the European starling (flexed-wing  
16 upstroke). We animate skeletal motion with XROMM (X-ray Reconstruction of  
17 Moving Morphology) to examine the contributions of each skeletal element of the  
18 wing: the contribution of the humerus, radius/ulna, and manus in turn. We find that  
19 both kinematic styles exhibit similar skeletal kinematic patterns, but differ  
20 significantly in magnitude and timing of joint excursion. Notable exceptions to the  
21 similarities include high long-axis rotation of the manus during upstroke in the  
22 pigeon, which contributes heavily to overall wing pronation, while the starling  
23 exhibits negligible long-axis rotation at the wrist. In contrast, higher pronation in  
24 the ulna/radius is apparent in the starling during upstroke. Together, these two

25 exceptions coupled with differences in timing and excursion angles explain variation  
26 in upstroke patterns.

27

28 **Keywords:** XROMM, bird flight, wrist, manus, long axis rotation, wingtip-reversal  
29 upstroke

30

31

## 32 **Introduction**

33

34 Birds capitalize on a suite of aerodynamic mechanisms to enhance force  
35 production during flight. During slow flight, birds are aerodynamically challenged,  
36 as most (if not all) of the induced velocity necessary to remain aloft and accelerate  
37 must be produced by the bird, rather than by redirecting existing airflow.  
38 Unsurprisingly, this phase is the most energetically costly phase of aerial locomotion  
39 for birds (Tobalske *et al.* 2003). To improve force production during this phase,  
40 birds with long, pointed wings use a ‘wingtip-reversal upstroke.’ These birds  
41 capitalize on a clap and fling mechanism during the transition from upstroke to  
42 downstroke (Crandell & Tobalske 2015) and induced momentum to the wake  
43 during mid-upstroke (Crandell & Tobalske 2011; Crandell & Tobalske 2015). In  
44 contrast, birds with rounded wings have a ‘flexed-wing upstroke,’ and do not  
45 capitalize on additional force production during the upstroke (Crandell & Tobalske  
46 2015), with the notable exception of the Phasianidae (Tobalske & Dial 2000).

47



48           The aerodynamic signature of both wingbeat styles appears closely tied to  
49 the external wing kinematics in Diamond doves and Zebra finch (Crandell &  
50 Tobalske 2015). However, the musculo-skeletal contribution to these motions  
51 remain unknown. Because birds that produce a flexed-wing upstroke have short,  
52 rounded wings, these wings exhibit proportionally less momentum during upstroke.  
53 It is possible that birds with different external kinematic patterns utilize similar  
54 skeletal kinematic patterns, but the resulting external wing motions are drastically  
55 different between upstroke styles, as momentum may dominate a long, pointed  
56 wing.

57           Avian upstrokes may be actively controlled, or a by-product of wing shape.  
58 Support for a muscularly-powered wingtip-reversal upstroke appears in both the  
59 muscle (Corvidae, Bierregaard & Peters 2006) and skeletal (Dial 1992b)  
60 morphology in comparative analyses. Specifically, birds that appear to use unsteady  
61 aerodynamic mechanisms more often have a different forelimb skeletal morphology.  
62 The ratio between humerus and ulna lengths correlates with three roughly defined  
63 classes of flight kinematics, from steady to unsteady (Nudds 2007). Similarly,  
64 forelimb skeletal elements appear to be more robust in species that utilize unsteady  
65 mechanisms. The radius and ulna bend away from each other, suggested to  
66 accommodate relatively large distal muscles (see Fig. 7 in (Dial 1992b)). Direct  
67 measures of musculature in three hawk species support this idea: Opsrey (*Pandion*  
68 *haliaetus*), a species observed with a pronounced tip-reversal upstroke during  
69 foraging, has a significantly larger supinator muscle than two other species  
70 (Corvidae, Bierregaard & Peters 2006), after accounting for between-species size

71 differences. Muscles that insert on the most distal skeletal elements, the manus,  
72 appear to follow the fine control of the hand-wing (Vazquez 1995). In fact, the  
73 pigeon is unable to take off and struggles in landing without the use of the  
74 antebrachial muscles (Dial 1992b). These anatomical observations suggest that  
75 direct muscular control of the distal wing is critical during acceleration and  
76 deceleration.

77         In contrast, (Vasquez 1992) suggests that upstroke patterns are a  
78 consequence of skeletal morphology and do not require the distal muscles. Vasquez  
79 proposes that a passive skeletal mechanism is responsible for the supination of the  
80 manus. From a morphological analysis of duck (*Anseriformes*) wrist bones, he  
81 outlines the following prediction for supination: First, the ulnocarpo-metacarpal  
82 ligament is brought under tension. Then, due to this tension, the ventral ridge of the  
83 manus slides along the ulnare, causing the wing to flick (Vasquez 1992). This  
84 proposed mechanism remains entirely passive, with negligible muscle input to  
85 produce the dramatic supination.

86

87         Here, we explore skeletal contributions to the two distinct avian upstroke  
88 styles, using an XROMM (X-ray Reconstruction of Moving Morphology) analysis. We  
89 describe movement along the six degrees of freedom for each of the major skeletal  
90 elements of the avian wing, in order to compare the general motions of the wingtip-  
91 reversal and flexed-wing upstroke style.

92

93

94 **Methods**

95

96 *Birds & Surgical Procedures*

97           Three captive-bred pigeons (*Columba livia*; Kate Davis, Raptors of the  
98 Rockies) and three wild-caught European starlings (*Sturnus vulgaris*; Fort Missoula  
99 Research Station, Missoula, MT) were used in this study. Pigeons were trained for  
100 two weeks prior to surgical procedure to cue in on a white platform to land.  
101 Starlings were trained for four weeks prior to procedures to cue in on a cylindrical  
102 (rod) perch at the end of a mesh tunnel. Reported herein are values for one pigeon  
103 (mass 262 grams) and three starlings (mass  $68.0 \pm 2.8$  grams).

104           Surgical procedures were done at the Field Research Station at Fort Missoula  
105 (Missoula, MT, USA). Birds were anesthetized using inhaled isoflurane (HME109,  
106 5% to induce, 2-3% to maintain; Highland Medical Equipment, Temecula, CA, USA).  
107 Feathers were removed as needed over the incision sites. At each incision site, the  
108 skin was manipulated to maximize access to a given location along the skeleton.  
109 Incisions were made at the keel, dorsal proximal end of the humerus, ventral distal  
110 end of the humerus (includes access to the proximal ulna), and the dorsal and  
111 ventral wrist. Implantation directly in to the distal carpometacarpus was possible  
112 without incision beyond the use of the drill. Skeletal implants consisted of 0.8 or 0.5  
113 mm diameter tantalum beads (BalTec, Los Angeles, CA). Implantation holes were  
114 drilled directly in to the bone with drill bits (of width 0.09 or 0.06 mm; McMaster  
115 Carr) mounted in insect pin holders. Incisions were closed with suture (0-3 or 0-4  
116 silk), or in instances of small incisions, vet-bond. Post-surgery birds were kept in a

117 small heated cages (food and water *ad libitum*) to recover for the first 6 hours, heat  
118 was removed after six hours. After 24 hours, birds were examined and moved back  
119 to aviaries. Birds were given one to two weeks recovery time before flight trials. All  
120 procedures approved by the University of Montana Institutional Animal Care and  
121 Use Committee (IACUC).

122

### 123 *Flight Trials*

124 Trials were video recorded at the W. M Keck Foundation XROMM (X-ray  
125 Reconstruction of Moving Morphology) Lab at Brown University (Providence, RI).  
126 Birds were flown between two Varian model G-1086 x-ray tubes and two Dunlee  
127 model TH9447QXH590 image intensifiers. Video outputs on the image intensifiers  
128 were recorded using two Phantom v10 high-speed cameras (1760x1760 pixels).  
129 Cameras were synced via a TTL pulse and sampled at 500 frames per second, with a  
130 shutter speed of 1/5,000 s. Both pigeons and starlings were flown through a  
131 constructed mesh corridor with a white platform (pigeons) or perch (starlings) at  
132 the end. To time releases to x-ray operation, birds were released by hand and flew  
133 two to three wingbeats before flying between intensifiers.

134 The three-dimensional interrogation volume was calibrated via standard  
135 XROMM Keck lab methods (xromm.org). Briefly, image distortion was accounted for  
136 by taking an image with a perforated metal grid taped to the image intensifier. The  
137 “undistortion” program (xromm.org) was used to create a transformation matrix.  
138 This matrix was applied to all subsequent images (calibration and trial).  
139 Calibrations were done using the Keck lab large cube.

140

141 *CT scans and skeletal models*

142           Following locomotor trials, all birds were euthanized. Bird carcasses were  
143 scanned at Brown University with the Keck Foundation CT-scanner (Fidex  
144 Veterinary CT scanner) (Starling slice thickness 0.199701, 80KvP, 43 mA; Pigeon  
145 slice thickness 0.216342, 80 KvP, 43 mA). Each bone was segmented out in Osirix  
146 software (32-bit) and saved as polygonal mesh models. Marker models for each  
147 bone segment were additionally saved as polygonal mesh models. Bone models  
148 were linked to marker models in Maya.

149

150 *Animation*

151           To measure the precision of our data, inter-marker distances were calculated  
152 for pairs of markers within the same bone (Figure 1, table 1). This approximation  
153 assumes negligible bone deformation during locomotion, and that the manus is  
154 treated as a unit (while it is composed of the carpometacarpus, phalanx 1 and  
155 phalanx 2). From 10 pairs of markers, the lowest standard deviation was 0.078 mm  
156 between point 1 and 4 of the trunk. The highest standard deviation was 0.17 mm  
157 between point 14 and 18 of the manus (Table 1). Notably, point 14 resides on the  
158 carpometacarpus, while point 18 was on the phalanx 3.

159           Marker coordinates from direct linear transformed data were filtered using a  
160 low-pass Butterworth filter with a 25 Hz cutoff frequency. Filtered markers were  
161 used to calculate rigid-body translations and rotations (see (Brainerd *et al.* 2010)).  
162 Rigid body transformations were applied to mesh bone models in Autodesk Maya

163 (2012, Autodesk Inc., San Rafael, CA, USA) using the XROMM package. One-marker  
164 rotoscoping was employed for the pigeon ulna. Two-marker rotoscoping was  
165 employed for all the starling ulnas and trunks.

166

### 167 *Data Analysis*

168 To describe motion around the wing joints, kinematic data were extracted in  
169 Maya using joint coordinate systems (Figure 2). The wrist joint represents motion of  
170 the carpometacarpus relative to the radius and ulna. The elbow joint represents  
171 motion of the radius and ulna relative to the humerus. For all joints, the radius and  
172 ulna were treated as a single fixed bone, when in reality slight motion between the  
173 radius and ulna may exist. Likewise, the manus was treated as one discrete element  
174 (including the carpometacarpus and phalanx 2 and 3). Additionally, the trunk was  
175 treated as a single solid object, and included the spinal column, pelvic, and pectoral  
176 girdles. This eliminated our ability to accommodate for changes in compression of  
177 the furcula – known to move up to 129% wider from resting length (Baier, Gatesy &  
178 Dial 2013). Our ‘zero’ position was selected as mid-downstroke posture - all  
179 reported values herein are relative to the bone positions in mid-downstroke (Figure  
180 2) for both species. In general, our placements of joint coordinate systems followed  
181 the conventions of Baier et al. (2013). Briefly, these systems follow the inertial axes  
182 of the bone, wherein the z axis (blue, Figure 2) experiences the highest inertia of the  
183 more distal bone in the joint, and the x-axis, the lowest. Thus, the z-axis generally  
184 followed the ab- and adduction plane (although at the shoulder, it followed flexion-  
185 extension), and the x-axis follows the long-axis rotation of the distal long bone. The

186 y-axis lies perpendicular to both. One notable exception to Baier et al. (2013) is that  
187 our study focuses on the long axis rotation of the long bones, an axis that is  
188 traditionally prioritized last (but see (Kambic, Roberts & Gatesy 2014)). Thus, our  
189 joint coordinate systems prioritized the long axis rotation axis over the ab- and  
190 adduction axis, which remains perpendicular to long-axis rotation.

191 Wingbeat timing was identified as being at the turn-around of the distal  
192 wingtip in both x-ray views. A slight delay between skeletal and feather turn-around  
193 may exist, and so we suggest interpreting comparisons with external kinematics  
194 with caution due to uncertainty about the precise upstroke-downstroke transition  
195 moment (see Baier, Gatesy & Dial 2013).

196 In order to measure how each joint contributes to the whole wing motion in  
197 both species, we modified the methods of Hedrick et al. (2012) and Baier et al.  
198 (2013). We created a tip trace for the distal-most point on the carpometacarpus,  
199 and tracked it throughout a complete wingbeat cycle, with coordinates centered at  
200 the shoulder joint. We then removed all motion due to the humerus, and then the  
201 radius/ulna. The traces remaining are motion due just to the motion of the bones  
202 distal to the targeted bone. For example, tip traces for the ulna include motion of the  
203 ulna and the manus, while tip traces for the manus include only motion of the  
204 manus.

205

## 206 **Results**

207 We found several differences between the pigeon and starling kinematics  
208 during a single wingbeat cycle. We find that the tip reversal upstroke is largely due

209 to pronation of the hand-wing, skeletally accomplished with long axis rotation of the  
210 manus at the wrist joint. In contrast, the flexed-wing upstroke in the starling is  
211 largely accomplished by flexion and long-axis rotation at the elbow joint.

212

### 213 *Wrist Joint Kinematics*

214

215         During the upstroke in the pigeon, a majority of the carpometacarpus (CMC)  
216 motion was due to the long axis rotation of the bone (Fig. 3A). During the upstroke,  
217 the wrist supinates, and then pronates during the wingtip-reversal upstroke.  
218 During the upstroke, the manus undergoes a long axis rotation of  $79.2^\circ$  (the  
219 difference between the maximum and minimum values during upstroke are  
220 reported herein). During the downstroke, long axis rotation is reduced, moving  
221 through  $49.4^\circ$ . A majority of that occurs as the manus continues to pronate at the  
222 transition between upstroke and downstroke. The manus then gradually supinates  
223 throughout downstroke.

224         In contrast, the starling manus goes through a smaller long axis rotation  
225 during upstroke, moving  $16.4^\circ$  (from  $-0.5 \pm 4.3^\circ$  to  $15.8 \pm 1.5^\circ$ ) during upstroke (Fig  
226 3D). A majority of this motion occurs early in the upstroke, where the CMC pronates  
227 before it is abducted (Fig. 3E). However, it is supinated, and during the last half of  
228 the upstroke (25-50% of the total stroke cycle) the wing undergoes negligible long-  
229 axis rotation.

230         The manus in the pigeon goes through minimal abduction during upstroke  
231 (up to  $34.0^\circ$  at the upstroke-downstroke transition), suggesting this motion is due to



232 a more proximal skeletal element (Fig 3E). In contrast, the starling manus is  
233 abducted heavily, abducting  $33.0^\circ$  (from  $-39.4 \pm 16.6^\circ$  to  $-6.4 \pm 7.5^\circ$ ) (Fig 3E) while  
234 the wing is flexed  $66.6^\circ$  (from  $-87.4 \pm 47.3^\circ$  to  $-20.8 \pm 23.2^\circ$ ) toward the body (Fig  
235 3F). The pigeon flexes the manus at the wrist  $47.4^\circ$  - consistent with the hand-wing  
236 remaining extended, but flexed toward the body.

237 In addition to the joint coordinate system, we qualitatively addressed the  
238 interaction between the carpometacarpus and the wrist bone, the ulnare (or  
239 cuneiform). We find visual confirmation that there is an interaction between the  
240 ventral ridge of the carpometacarpus and the ventral ramus of the ulnare during the  
241 tip-reversal portion of the upstroke (25-50% of wingbeat cycle; Supplemental Video  
242 1).

243

#### 244 *Elbow Joint Kinematics*

245

246 During the upstroke in the pigeon, the ulna pronates  $44.9^\circ$  (Fig 4A) as the  
247 manus supinates (Fig 3A). Before downstroke, the ulna supinates back to end of  
248 downstroke values. The starling follows a similar pattern, pronating  $92.3^\circ$  during  
249 the first half of upstroke (0-25% of wingbeat cycle; from  $8.3 \pm 28.5^\circ$  to  $84.0 \pm 42.1^\circ$ ).  
250 During downstroke, the pigeon does not go through much long-axis rotation at the  
251 elbow. Over the course of the entire downstroke (50-100% of the wingbeat cycle),  
252 the starling supinates through  $50.3^\circ$  (from  $38.7 \pm 23.9^\circ$  at the start of downstroke to  
253  $-11.6 \pm 8.2^\circ$  at the end of downstroke).

254 In both the pigeon and starling, the elbow abducts through the first half of  
255 upstroke (Fig 4B, 4E) (0-25% wingbeat cycle), and then adducts for the second half  
256 (25-50% of the wingbeat cycle), prior to downstroke. The starling goes through an  
257 arc ( $28.8^\circ$  - from  $36.2 \pm 22.6^\circ$  to  $7.4 \pm 7.7^\circ$ ) approximately that of the pigeon ( $45.5^\circ$ ).  
258 During downstroke, the elbow joint remains constant – neither abducting nor  
259 adducting heavily for both birds.

260 Levels of flexion and extension of the elbow joint (Fig. 4C, F) are comparable  
261 for the pigeon and starling. The starling flexes and extends the ulna and radius  
262 between  $-95.6 \pm 44.6^\circ$  and  $-40.1 \pm 35.2^\circ$ , a path of  $55.6^\circ$ . The pigeon flexes from  $3.5^\circ$   
263 to a maximum of  $35.1^\circ$ , a total arc of  $38.5^\circ$ .

264

### 265 *Shoulder Joint Kinematics*

266

267 During the upstroke, the pigeon humerus supinates  $98.3^\circ$  rapidly at 25-30%  
268 through the wingbeat cycle (Figure 5A). This occurs at the same time as the  
269 carpometacarpus begins to pronate, although during a shorter duration (pronation  
270 occurs approximately 20% of the wingbeat cycle). Following this supination, the  
271 humerus remains heavily supinated throughout the ‘clap’ and in to the ‘peel’  
272 upstroke-downstroke transition. For the starling, a similar pattern of long axis  
273 rotation (Figure 5D), to a lesser degree, occurs during the first 33% of the wingbeat  
274 cycle, supinating through  $44.7^\circ$  - from  $-22 \pm 16.7^\circ$  to  $-67.6 \pm 28.1^\circ$ .

275 During this rapid supination in the pigeon, the humerus also abducts  $146.1^\circ$   
276 quickly to extend the wing dorsally, setting the bird up for the ‘clap and peel’

277 upstroke to downstroke transition (Figure 5B). In the starling, a similar pattern of  
278 long axis rotation, covering  $130.1^\circ$ , from  $-43.6 \pm 21.2^\circ$  to  $87.3 \pm 43.9^\circ$  occurs over a  
279 much longer time interval (approximately 40% of the wingbeat cycle).

280         The pigeon humerus flexes toward the body throughout the initial 33% of  
281 the upstroke, flexing  $85.5^\circ$  (Fig. 5C). In contrast, the starling flexes toward the body  
282 at a much lower angle,  $44.9^\circ$  (Fig 5F; from  $-12.7 \pm 10.6^\circ$  to  $-57.6 \pm 37.3^\circ$ ).

283

#### 284 *Joint contributions to whole-wing motion*

285

286         In the pigeon, we found that the shoulder is responsible for a majority of the  
287 cranial-caudal motion of the wingtip (86% of wingbeat, Figure 6B, E). In contrast,  
288 the elbow contributes most to dorso-ventral motion of the wing (94% of wingbeat,  
289 Figure 6C). The wrist joint contributes most to motion in the lateral plane (68%,  
290 Figure 6G).

291         For the starling, the shoulder is the dominant joint responsible for cranial-  
292 caudal motion (48% of all motion centered at the shoulder), as well as dorso-ventral  
293 motion (44%). The wingtip-trace of isolated wrist motion follows the same general  
294 pattern (to a lesser degree) as the shoulder joint in the lateral view. However, the  
295 amount of flexion is much reduced at the wrist – indicating that the flexion of the  
296 elbow joint is crucial for the flexed-wing upstroke. In fact, the elbow joint dominates  
297 all flexion of the wingtip (Figure 7E).

298

299

300 **Discussion**

301

302           We have established that the skeletal kinematics in birds during slow flight is  
303 different between the two upstroke styles. Although the skeletal elements have the  
304 same general kinematic motions at each wing joint, the timing and excursion angles  
305 differ, and so, are responsible for the different external wing kinematics. In the  
306 pigeon, the wingtip-reversal upstroke is largely dominated by high long-axis  
307 rotation ( $80^\circ$ ) at the wrist mid-way through the upstroke, and quick long-axis  
308 rotation of the humerus ( $98^\circ$ ) near the end of upstroke. In contrast, the flexed-wing  
309 upstroke of the starling is dominated by flexion of the ulna at the elbow joint.

310           One notable exception to all joints undergoing similar kinematic trajectories  
311 is in the long-axis rotation of the wrist. The starling wrist undergoes negligible long-  
312 axis rotation. The rotation that it does go through is in the opposite direction to that  
313 of the pigeon: the manus of the starling pronates slightly as the pigeon manus  
314 supinates at the start of the upstroke. This one difference appears crucial for the  
315 tip-reversal upstroke of the pigeon – in particular, the pronation of the primary  
316 feathers during the upstroke- to downstroke transition.

317           We find qualitative evidence that supports the hypothesis presented by  
318 Vazquez (1992), wherein pronation of the hand-wing during the tip-reversal  
319 upstroke is due to an interaction between the carpometacarpus and the ulnare  
320 (supplemental video 1). This video was accomplished with the assumption that the  
321 ulnare follows the general kinematic trajectory of it's more proximal element, the  
322 ulna/radius, and so is an approximation of wrist bone kinematics. With these

323 assumptions acknowledged, we observe that the carpometacarpus slides along the  
324 ventral ramus of the ulnare. Further work to explore the relative contribution of the  
325 distal muscular elements to this motion is necessary to validate Vazquez' (1992)  
326 hypothesis that this motion is muscularly passive, and completely skeletally-driven.  
327 Additionally, higher resolution studies with more wrist bone markers are necessary  
328 to validate our assumption of ulnare motion.

329         Our kinematic observations for both birds are different from those reported  
330 in the hummingbird. In hummingbirds, the wing inversion during transitions from  
331 upstroke to downstroke and vice-versa is predominately due to action at the  
332 shoulder joint (Hedrick *et al.* 2012). However, in both species studied here, much of  
333 wing motion occurs in the more distal joints (elbow, wrist), as opposed to the  
334 shoulder joint. Distal motion does rely on the humerus (Fig. 5, 6), but is not the sole  
335 contributor to movement. In particular, long-axis rotation of the humerus occurs  
336 over a very short period of time at the end of upstroke in the pigeon. This long axis  
337 rotation was first observed in both pigeons and starlings based on *in vivo* muscle  
338 stimulation studies (Poore *et al.* 1997), and from rotoscoping of the pigeon humerus  
339 (Gatesy *et al.* 2010). This motion likely contributes to reorienting the secondary  
340 feathers dorsally during an aerodynamically active “clap” phase, wherein the left  
341 and right wing come in to contact and shed circulation from the wing (Crandell &  
342 Tobalske 2015).

343         We found evidence that the skeletal elements of the bird wing do not lock in  
344 to place during the downstroke –first proposed by Sy in 1936 (Sy 1936; Vasquez  
345 1992). In particular, we observed significant motion in all degrees of freedom in

346 both species throughout the downstroke (Fig. 3, 4, 5). Examining wingtip-traces  
347 once the effects of the most proximal element (humerus) are removed easily reveals  
348 that a substantial amount of abduction, adduction, flexion, and extension occur at  
349 the more distal joints. Additionally, substantial long-axis rotation is observable  
350 during the downstroke in both the wrist of the pigeon (Fig. 3A) and elbow of both  
351 species (Fig. 4A,D). This long-axis rotation may be due to aerodynamic loading of  
352 the feathers, as it is in the proper direction if the wing is aerodynamically loaded  
353 with the center of pressure behind the bones. If this is so, differences in the degree  
354 of long-axis rotation may be due to smaller distal muscle masses in the starling, and  
355 so less muscle work available to maintain proper wing orientation. Further work  
356 examining distal muscle contribution to wing shape is necessary to elucidate the  
357 role of the distal muscles relative to aerodynamic loading of the wing.

358 Further, timing and magnitude of wrist extension varies relative to elbow  
359 extension, countering the idea of an automatic linkage system between limb  
360 segments in bird wings (Vazquez 1994). In the pigeon, the elbow begins extending  
361 approximately 10 milliseconds (or 8% of the length of a wingbeat cycle) before the  
362 wrist extends in the pigeon (Fig. 3C, 4C). In the starling, the delay is 4% of the  
363 wingbeat cycle<sup>4</sup> between the initiation of elbow extension and wrist extension. In  
364 the pigeon, the humerus begins extending after the extension of the elbow (11%)  
365 and wrist (5%). Baier et al. (2013) have shown similar timing differences in chukar  
366 partridges during ascending flight, but find timing similarities of limb extension in  
367 chukar during wing assisted incline running. Here, we add that flexion distally in the  
368 elbow and wrist occur before flexion at the humerus. This timing difference in the

369 opposite direction that would be expected if the linkage system is lead by  
370 movements of proximal skeletal elements (i.e., the humerus). Together, this suggests  
371 that an “automatic linkage system” is not used during slow flight. Such a system may  
372 exist and be deployed during fast flight or gliding, wherein maintaining a relatively  
373 flat wing is crucial to force production. Further work examining the skeletal system  
374 under different flight speeds is necessary.

375

376 Our data reveal that there are stereotypical patterns of motion regardless of  
377 external wing kinematics. However, these patterns vary between species in terms of  
378 the excursion angles and timing of actions. As skeletal kinematic patterns are similar  
379 across species, this suggests that the underlying muscles of the wing may operate  
380 under a central pattern generator (Brown 1911; Kiehn 2006), with muscle  
381 activation timing and magnitude modified over evolutionary time. Alternatively,  
382 muscles may have been co-opted to perform different locomotor tasks. Further  
383 work examining precise muscular control of wing morphing is necessary to  
384 elucidate the underlying control mechanisms (Dial 1992a).

385 Kinematic data illustrate that external wing (and feather) kinematics are  
386 driven by underlying skeletal motion, and not just passive aerodynamic or other  
387 kinetic aspects, such as inertia of the wing. The wingtip-traces for both species (Fig.  
388 6B, 7A), largely reflect the three-dimensional wingtip traces of the distal-most point  
389 of the 10<sup>th</sup> (or 9<sup>th</sup>) primary feather previously reported. To compare feather- to  
390 skeletal- motion, we compare our data to several sources of external kinematic data.  
391 Three-dimensional kinematics for the pigeon have been previously reported (Berg

392 & Biewener 2008; Berg & Biewener 2010) alongside two-dimensional kinematics of  
393 the pigeon (Tobalske & Dial 1996). Tobalske & Dial (1996) have also reported two-  
394 dimensional kinematic data for the black-billed magpie, which closely matches  
395 three-dimensional kinematic data reported for the Zebra finch in slow flight  
396 (Crandell & Tobalske, 2015). Thus, we assume for the purposes of this discussion  
397 that passerines (including the magpie, zebra finch, and the starling studied here)  
398 have a characteristic wingstroke during slow flight.

399         When comparing our skeletal data to sources of external wing kinematics,  
400 both pigeons and starlings include a higher abduction of the manus in comparison to  
401 the feather trace. Intriguingly, the manus in the pigeon follows along an identical  
402 lateral path during upstroke and downstroke. Feather tip-traces of a diamond dove  
403 (and pigeon) show differences in the degree of flexion toward the mid-line of the  
404 body, wherein the distal primary feather follows a figure-of-eight pattern. The  
405 feather tip stays closer to the mid-line of the body during the first half of upstroke in  
406 comparison to the wing tip trace during downstroke, while the feather tip is farther  
407 from the body during the end of upstroke in comparison to the start of downstroke  
408 (Figure 1 of Chapter 2; Crandell & Tobalske, 2015). We posit that the difference  
409 between the skeleton and feather observed kinematics may be due to feather inertia  
410 (Gatesy & Baier 2005) or aerodynamic loading (Chang *et al.* 2013; Crandell &  
411 Tobalske 2015).

412         Here, we have established that external wing kinematics of birds in slow  
413 flight are closely mirrored by internal skeletal kinematics. Further, all birds undergo  
414 similar patterns of motion skeletally during slow flight. However, differences in



415 magnitude of joint excursion and timing contribute greatly to overall differences in  
416 wing motion.

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437

438 **Figures & Tables**

439

440

441

Trunk:				
Marker #	2	3	4	
1	0.096	0.084	0.078	

Humerus:				
Marker #	6	7	8	
5	0.092	0.17	0.0258	

Manus:				
Marker #	15	16	17	18
14	0.12	0.164	0.116	0.17

442

443 **Table 1:** Standard deviations (mm) for each marker pair on the same bone for

444 pigeon WG during trial 7.

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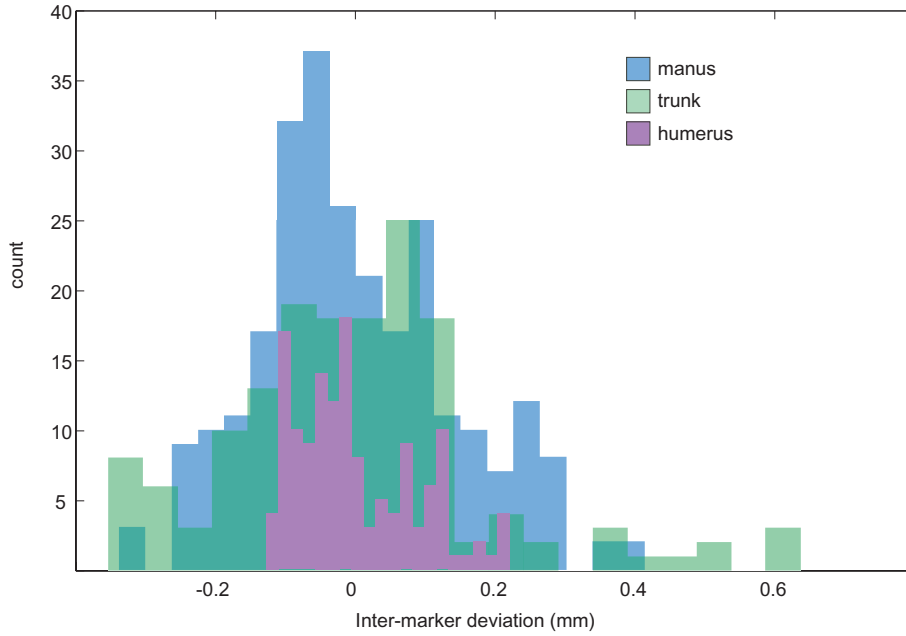
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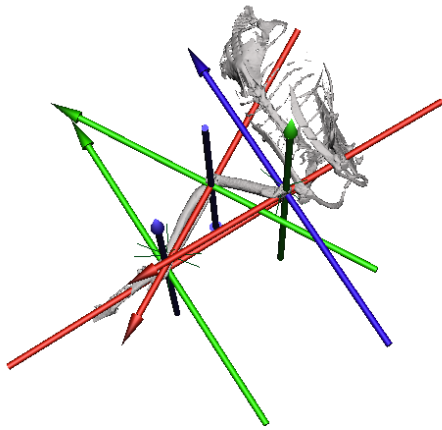
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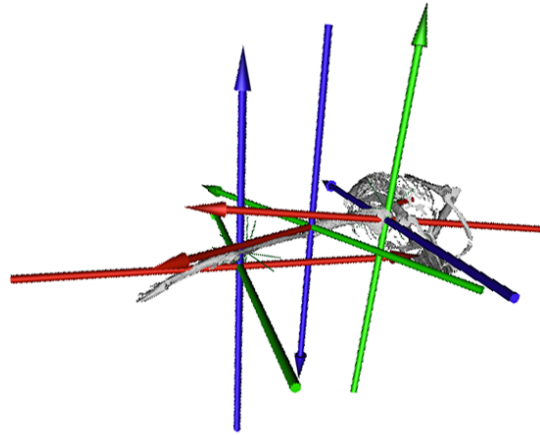
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**Figure 1:** Histogram of deviations from the mean distance (0) between two markers on the same bone for pigeon WG, run 7. Blue histogram includes data from four marker pairs on the manus. The green histogram includes 4 marker pairs on the trunk, and purple includes 3 marker pairs of the humerus.

(A) Dorsal-oblique view



(B) Cranial-oblique view



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482 **Figure 2:** (A) Dorsal-oblique and (B) Cranial-oblique view of the Joint Coordinate  
483 System (JCS) for all species, shown on the starling. Herein, all reported values are  
484 relative to this mid-downstroke position (the 'zero position'). Each joint has six  
485 degrees of freedom based on inertial axes of the distal bone. Degrees of freedom are  
486 as follows for the wrist and elbow joint: Blue = abduction/adduction, Green =  
487 flexion/extension, Red = supination/pronation. For the Shoulder joint, blue =  
488 flexion/extension, green = abduction/adduction, red= supination/pronation

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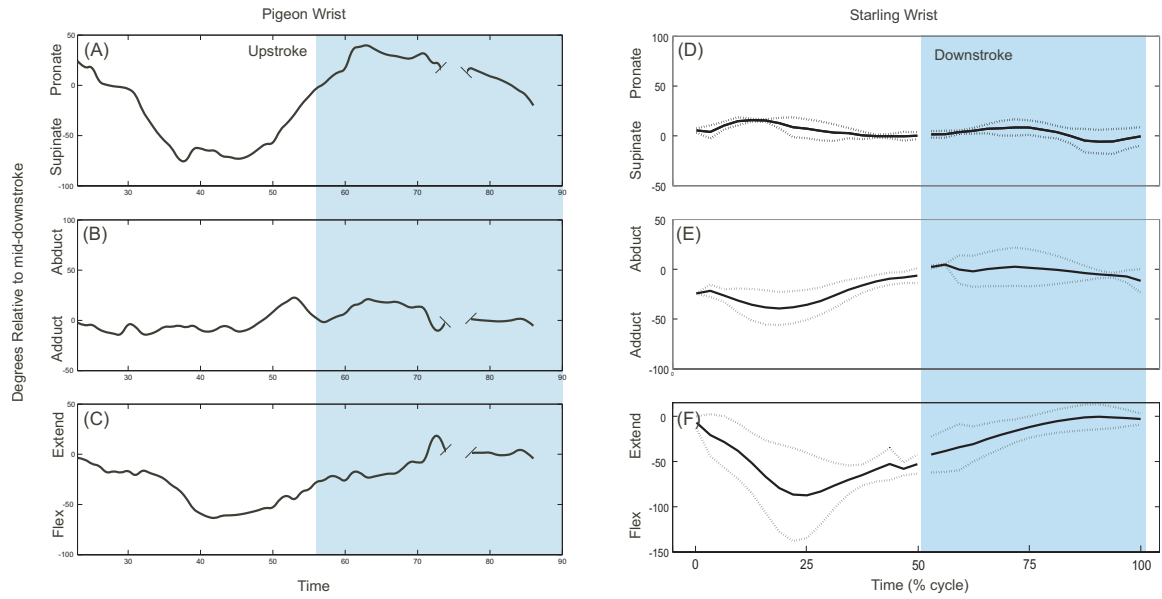
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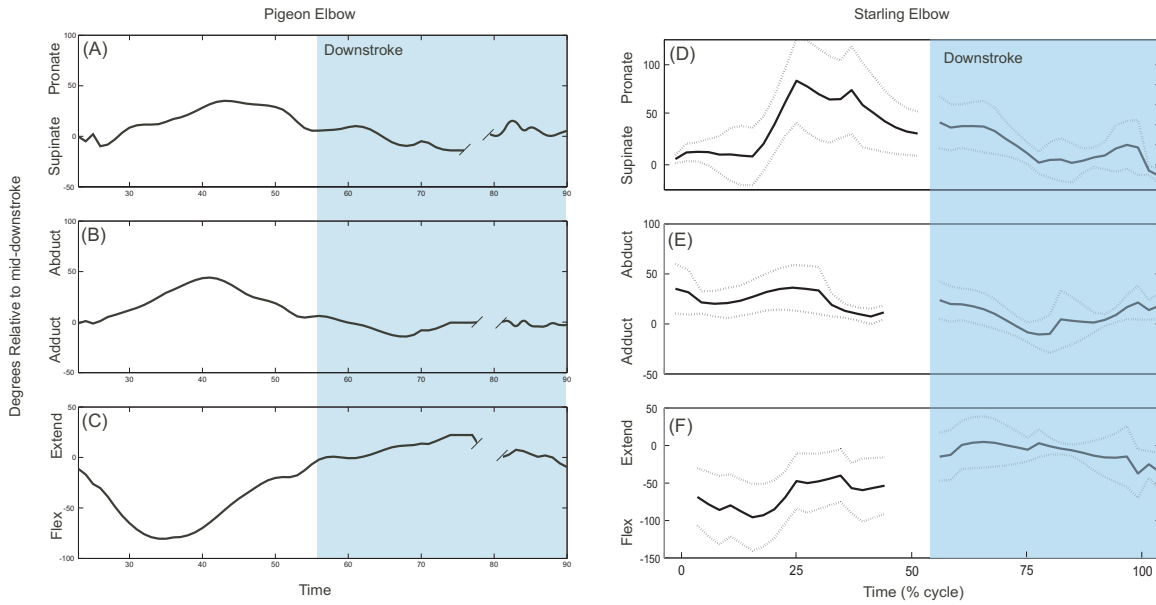
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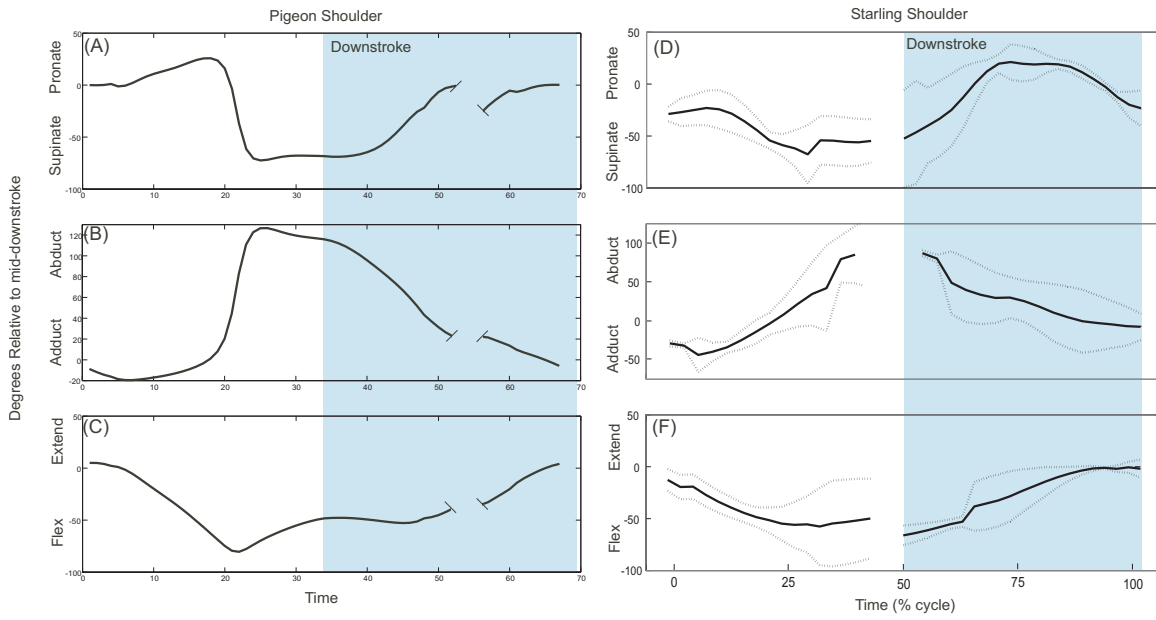
**Figure 3:** Representative trial for pigeon (A-C) and average (solid)  $\pm$  s.d. (dashed) for starlings ( $n=3$ ) (D-F). Each rotational degree of freedom for the wrist is shown. Line breaks in pigeon and starling represent missing data from stitching together of non-consecutive wingbeat segments.



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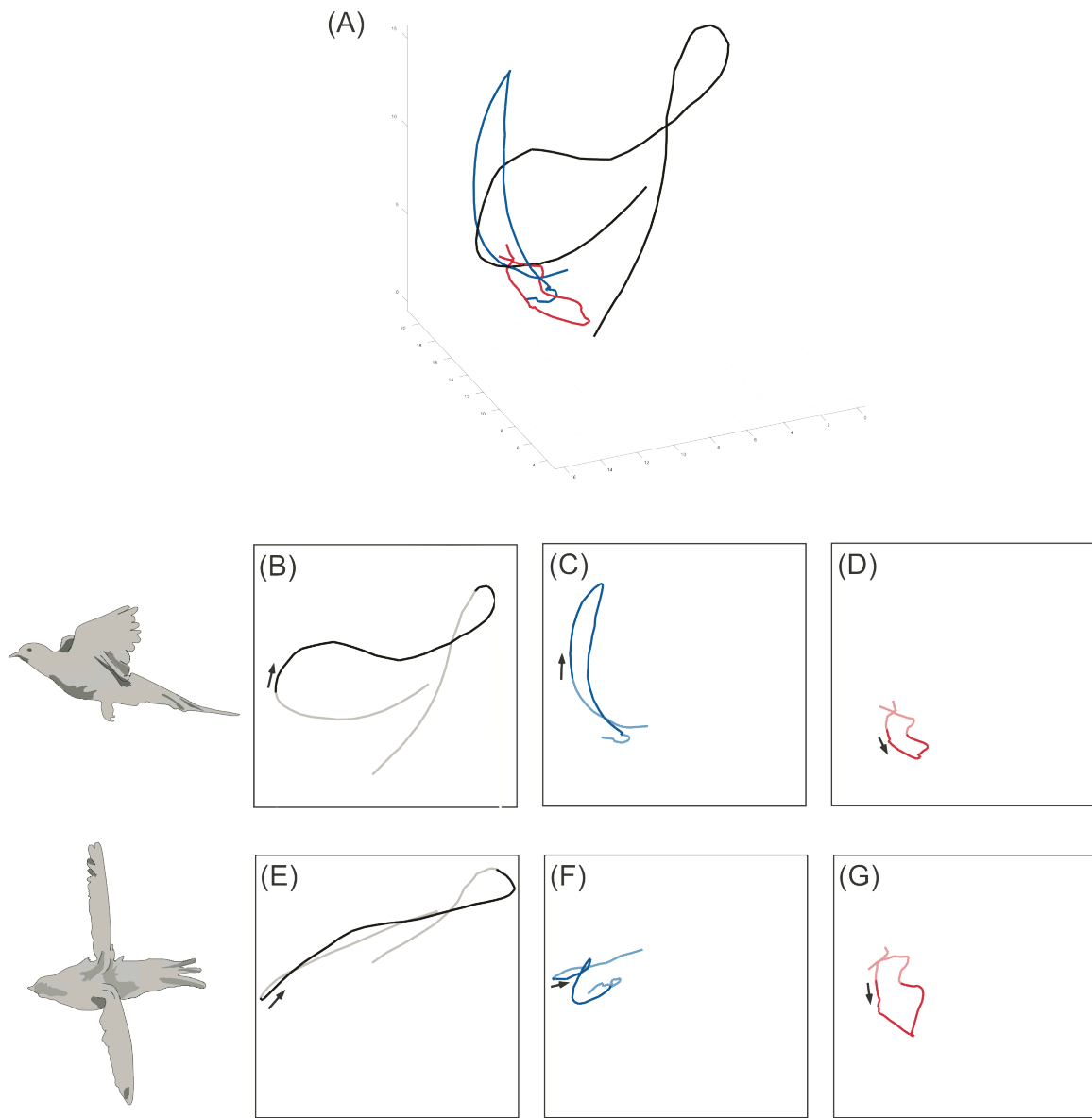
**Figure 4:** Representative trial for pigeon (A-C) and average (solid)  $\pm$  s.d. (dashed) for starlings (n=3) (D-F). Each rotational degree of freedom for the elbow is shown.





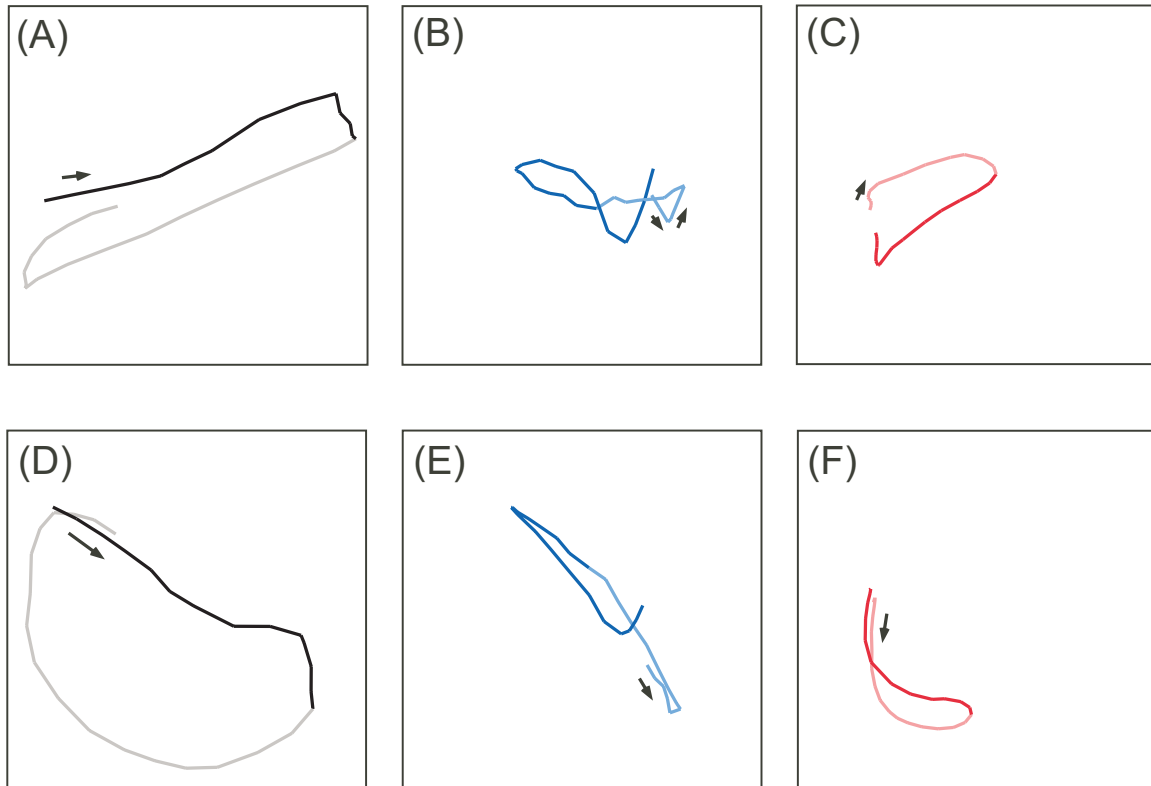
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**Figure 5:** Shoulder joint coordinate kinematics for a representative wingbeat cycle for a pigeon (A-C) and average (solid)  $\pm$  s.d. (dashed) for starlings (n=3) (D-F).



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**Figure 6:** Traces of the distal-most point on the manus (phalanx 3) during a representative wingbeat cycle in the pigeon. Lighter colors represent the downstroke phase, darker colors represent the upstroke. Black arrows indicate the start of upstroke and the direction of wingtip motion. (A) illustrates a three-dimensional view of the wingtip path with each trace representing the motion due to different skeletal elements. (B-D) is a lateral view of the tip traces, and (E-G) is a dorsal view. For all graphs, black = shoulder-centered coordinates of the wingtip (including motion of the humerus, radius/ulna, and carpometacarpus), blue = elbow-centered coordinates (including motion of the radius/ulna and carpometacarpus), red = wrist-centered coordinates (including only motion of the carpometacarpus).



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555 **Figure 7:** Traces of the distal-most point on the manus (phalanx 3) during a

556 representative wingbeat cycle in the starling, flying to the left. Lighter colors

557 represent the downstroke phase, darker colors represent the upstroke. Black

558 arrows indicate the start of upstroke and the direction of wingtip motion. (A)

559 illustrates a three-dimensional view of the wingtip path with each trace

560 representing the motion due to different skeletal elements. (A-C) is a lateral

561 view of the tip traces, and (D-F) is a dorsal view. For all graphs, black =

562 shoulder-centered coordinates of the wingtip (including motion of the humerus,

563 radius/ulna, and carpometacarpus), blue = elbow-centered coordinates

564 (including motion of the radius/ulna and carpometacarpus), red = wrist-

565 centered coordinates (including only motion of the carpometacarpus).

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578

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580 toward XROMM, we thank Ashley Heers, Andy Biewener, Beth Brainerd, Bret  
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## Chapter 4

### Coping with compliance during take-off and landing in the diamond dove (*Geopelia cuneata*)

Kristen E Crandell<sup>\*1</sup>, Austin F. Smith, Ondi Crino<sup>2</sup>, Bret W. Tobalske<sup>1</sup>

1. Field Research Station at Fort Missoula, Division of Biological Sciences, University of Montana, Missoula, MT USA.

2. Centre for Integrative Ecology, Deakin University, Victoria, Australia

\*author for correspondence ([kristen.crandell@gmail.com](mailto:kristen.crandell@gmail.com))

#### Abstract

Structural variation in natural habitats imposes constraints on locomotor performance. Birds appear to transition smoothly from aerial to terrestrial environments during take-offs and landings, however, these transitions are likely challenged by perch compliance (the opposite of stiffness). Given the ecological ubiquity of substrate variation, here we examine how birds modulate their performance during take off and landing on substrates of low, medium, and high compliance. We find birds do not capitalize on perch compliance, and so in general suffer from inefficient transitions. In particular, take-off velocities are negatively impacted by perch compliance, and landing velocities remain unchanged. Legs and wings function as independent units, such that lower initial flight velocity due to the legs negatively impacts later flight velocities. During landing, significant stability problems arose with compliance that were ameliorated by the wing and tail modules. Lastly, we find evidence for wild birds behaviorally avoiding the negative impacts of compliance by using perches of larger diameter.

### 33 **Introduction**

34

35           Natural habitats are full of structural variation that imposes constraints upon  
36 locomotor performance. Vegetation varies in orientation, roughness, size, and  
37 compliance. For example, the flexural stiffness of branches varies 4-fold within a  
38 single genus of trees in a single forest (van Casteren *et al.* 2013). Such variation can  
39 directly impact locomotor capacity. As an example, a slight change in the angle of a  
40 substrate (from 0 to +10 degrees) caused a 46% decrease in running velocity in  
41 geckos that had their adhesive system deployed (Russell & Higham 2009).

42           Substrate compliance (i.e., flexibility, or the opposite of stiffness) poses a  
43 unique challenge. As an organism moves, it transfers kinetic energy from itself to  
44 the substrate, which is stored as potential energy. Unless that potential energy is  
45 recovered, this results in lower sum energy for the animal to engage in the pendular  
46 mechanics of walking or effectively launch from the substrate during running and  
47 leaping (Alexander 1991). Shape, diameter, and length play a role in the compliance  
48 of substrates (Vogel 2003). Given the high degree of variation, it is reasonable to  
49 predict that organisms have adapted a diverse set of strategies to accommodate  
50 structural compliance within the habitat.

51           Response to substrate compliance appears to vary among species.  
52 Orangutans, for example, are capable of capitalizing on whole-tree compliance by  
53 using trees as an “external spring:” converting the potential energy of the tree into  
54 kinetic energy to move across the canopy (Thorpe, Crompton & Alexander 2007).  
55 However, in *Anolis* lizards, compliant perches constrain initial jump velocity



56 (Gilman *et al.* 2012). Anoles appear to behaviorally adjust with preferences toward  
57 stiffer substrates (Gilman & Irschick 2013).

58         Reactions to compliance may differ between terrestrial and volant  
59 organisms. Shifting between a terrestrial and aerial environment appears seamless  
60 for many flying birds, yet is understood to be expensive in terms of power (Tobalske  
61 *et al.* 2003). In birds, take-off and landing offer instances of high selective pressure.  
62 In many species, escape from predation, mate courtship, and foraging rely on an  
63 accurate take-off. A precise and rapid landing is also crucial to avoid wing or whole-  
64 body damage and to minimize energy expenditure. Thus, the logical prediction that  
65 follows is that birds should take-off and land as quickly as possible to minimize  
66 energy expenditure and avoid damage.

67         To maximize safety and minimize energetic costs during transitions between  
68 substrates and air, it was formerly thought that anatomical and physiological wing  
69 and leg modules in birds, which operate independently during terrestrial and aerial  
70 locomotion, instead function in tandem during transition stages (Gatesy & Dial  
71 1996). However, recent work has established that during take-off, the legs and  
72 wings continue to function discretely in certain species. The legs contribute most to  
73 initial acceleration in take-off, and the wings follow with significant forces only after  
74 the feet leave the substrate (Provini *et al.* 2012). During landing, the wings  
75 contribute the majority of the deceleration prior to leg interaction with the  
76 substrate (Provini *et al.* 2014). Control requirements during these phases are  
77 potentially challenged by large variation in compliance of natural perches. Little is  
78 known concerning how birds accommodate such a variety of structures during

79 transitions in spite of how this ties together the ecology of these animals with their  
80 primary locomotor style.

81         During take-off, birds do not appear to adjust their leg forces to  
82 accommodate variation in perch compliance (Bonser, Norman & Rayner 1999).  
83 However, two other adjustments may occur to maximize take-off velocity. The first  
84 is that birds may adjust the timing over which the force is applied. As velocity =  
85 (force \* time) / mass, if force remains constant, the remaining variable that can be  
86 adjusted for greater velocity is time. In contrast, volant organisms may compensate  
87 for lost performance from the hindlimbs in flight. Birds are known to modulate their  
88 force production following take-off by engaging their wings to different degrees  
89 (Provini *et al.* 2012). Therefore, even if energy is lost to the perch, causing a lower  
90 initial jump velocity (as suggested in European Starlings; (Bonser, Norman & Rayner  
91 1999)), volant organisms may be able to adjust in flight.

92         During landing, the leg forces are smaller than those during take-off (Provini  
93 *et al.* 2012; Provini *et al.* 2014). In comparison to terrestrial organisms, aerial  
94 landings capitalize on aerodynamic decelerations prior to landing, and so impact  
95 forces are lower. As such, volant organisms may be able to prepare prior to landing  
96 to accommodate variation in substrate properties. Pigeons approach a novel perch  
97 at a much lower flight velocity than a well-known perch (Green & Cheng 1998),  
98 suggesting that variation in preparation for landing on known substrates occurs in  
99 birds.

100         Given the ecological significance of compliant substrate use and the  
101 differences between the forces produced during take-off and landing in terrestrial

102 and aerial organisms, we decided to examine how perch compliance affects multiple  
103 aspects of take-off and landing in the diamond dove (*Geopelia cuneata*). Here, we  
104 address the following questions: (1) How are take-off and landing velocities  
105 impacted by perch compliance? (2) Do birds coordinate leg and wing modules to  
106 accommodate perch compliance? (3) Do birds select perches to avoid or enhance  
107 potential effects of compliance?

108

109

## 110 **Methods**

### 111 *Perch Use*

112 Five diamond doves (*Geopelia cuneata*) were trained to fly horizontally in the  
113 laboratory between two perches two meters apart. During training, birds randomly  
114 experienced each of three perch types daily for two weeks prior to data collection.

115 During experiments, perches were mounted to a custom 3-axis force plate  
116 (Bertec). We varied perch compliance (the opposite of flexural stiffness) using wood  
117 (flexural stiffness,  $EI=0.39 \text{ Nm}^2$ ), aluminum ( $EI=2.12 \text{ Nm}^2$ ), or steel ( $EI = 6.14 \text{ Nm}^2$ )  
118 perches, each 5 mm in diameter. Estimated deflection for these perches, with a 50 g  
119 bird perched at the takeoff location, 40 cm from the point of attachment, were 2.6  
120 cm, 0.5 cm, and 0.2 cm, respectively. During recorded trials, birds were given five to  
121 ten practice transitions on each substrate prior to data recording, intending to re-  
122 familiarize them with the substrate. Ten trials were collected per substrate, per bird,  
123 for both take-off and landing (60 trials total per bird). The order substrates were

124 used was randomized for each bird, but to allow for any known learning effects,  
125 perch type was not changed between the ten trials on each substrate.

126

127         From the force traces, we extracted resultant velocities ( $\text{m s}^{-1}$ ), peak forces  
128 (N), and time (s) of force application. Time of force application was measured  
129 between the time after countermovement (when vertical force is less than body  
130 weight (Earls 2000; Provini *et al.* 2012)) and the time the bird's feet were off the  
131 perch, confirmed visually in high-speed video. During landing, low-frequency  
132 oscillations made suspect any measurements of instantaneous forces, but measuring  
133 timing to stability was feasible.

134         To measure wing contributions to accelerations and decelerations in body  
135 velocity, a high-speed video (Photron 1024 PCI) was taken laterally at 1,000 fps  
136 (shutter speed = 1/10,000). To test for wing recovery following takeoff, we  
137 measured flight velocity during the third complete wingbeat of flight. To test for  
138 potential wing use variation between substrates in preparation for landing, we also  
139 measured the flight velocity during the final complete wingbeat before landing.

140

#### 141 *Habitat Use Measurements*

142         To measure perch selection versus availability in the wild, birds were  
143 observed at Philip Creek Cattle Station (Philip Creek, Northern Territory, Australia),  
144 and Burt Plain Cattle Station (Tanami Road, Alice Springs, Northern Territory,  
145 Australia). Doves were located with in response to visual and auditory cues. When a  
146 perched diamond dove was first encountered, the following information of the

147 initial sighting was recorded: perch diameter and distance to fulcrum (determined  
148 as distance to an intersecting branch/trunk >5 cm in diameter, or the visually-  
149 identified main stalk of the plant). To compare bird sightings to available habitat, 10  
150 photographs were taken of surrounding habitat, and diameters and distances to  
151 fulcrum were measured in ImageJ for 200 randomly selected points.

152

### 153 *Data analysis*

154 To test for differences in performance parameters between substrate type,  
155 we used a within-subjects, repeated-measures ANOVA, with individual bird as a  
156 factor. To compare available vs. used habitat type, we used a t-test.

157

## 158 **Results**

159 We found that performance varied with perch compliance during both take-  
160 off and landing.

### 161 *Take-off*

162 Take-off velocities were lower with more compliant perches. Takeoff  
163 resultant velocities were significantly greater on the perch with the lowest  
164 compliance, steel (Figure 1A; ANOVA within-subjects factor  $F_{2,8}=22.26$ ,  $p<0.01$ ).  
165 Resultant velocities produced during take-off from aluminum and wood perches  
166 were less, and not statistically different from each other. Birds produced a velocity  
167 of  $0.50 \pm 0.12 \text{ ms}^{-1}$  taking off from wood, followed by  $0.53 \pm 0.11 \text{ ms}^{-1}$  from the  
168 aluminum perch, and  $0.82 \pm 0.21 \text{ ms}^{-1}$  from the steel perch. We did not find that  
169 timing of force application changes with substrate compliance, indicating that birds

170 do not adjust force or timing of force application to improve take-off velocity in  
171 response to perch compliance. However, a general trend toward longer force  
172 application times was apparent, although not statistically significant ((Figure 1B;  
173  $F_{2,8}=1.89$ ,  $p=0.21$ ; wood =  $0.05 \pm 0.02$  s, aluminum =  $0.06 \pm 0.01$  s, steel =  $0.07 \pm$   
174  $0.02$ ). Additionally, we did not find evidence that birds utilized perch recoil to  
175 recover potential energy from the perch as kinetic energy. In all instances, the bird  
176 lost contact with the perch prior to the perch recoiling (Supplemental Video 1).  
177 Birds did appear to compensate for velocity lost on compliant substrates during  
178 take-off in flight. However, by the third wingbeat in flight, velocity lost via the legs  
179 due to perch compliance remained statistically significant: doves taking off from a  
180 wood perch had an in-flight velocity of  $1.40 \pm 0.34$  ms<sup>-1</sup>, take-off from an aluminum  
181 perch had a flight velocity at  $1.46 \pm 0.26$  ms<sup>-1</sup>, and following steel at  $1.59 \pm 0.25$  ms<sup>-1</sup>  
182 (Figure 1C;  $F_{2,8} = 12.04$ ,  $p<0.01$ ).

183

#### 184 *Landing*

185 During landing, we found that birds approach perches at constant velocities,  
186 with statistically non-significant differences ( $F_{2,4} = 1.36$ ,  $p=0.35$ ), regardless of perch  
187 compliance (Figure 2). This negates the idea that birds prepare for landing to  
188 capitalize on the physical properties of known substrates. In contrast, we found that  
189 stabilization and control were greatly challenged on compliant substrates during  
190 landing.

191 Upon landing, we saw predictable patterns of body reorientation. At touch-  
192 down, the bird immediately adjusts to a more upright posture, and then folds the

193 wings – as observed upon landing on the substrate with minimal-compliance, steel  
194 (Figure 3A, steel). During this reorientation phase, the bird initially touches down at  
195 a body angle between the eye and rump of  $26.3 \pm 45.8^\circ$  from horizontal. The bird  
196 then reorients to a resting angle of  $70.4 \pm 2.8^\circ$  from horizontal. During the  
197 reorientation phase, the bird peak angular velocity of reorientation occurs at 50  
198 milliseconds after touch-down, at  $11.4 \pm 3.3 \text{ rad s}^{-1}$ .

199         A failure in stability, then, can be quantified when this reorientation is  
200 delayed or variable. In both wood and aluminum treatments, birds took  
201 considerably longer to reach stability, and showed high standard deviations in the  
202 timing of stability (Figure 3B and 3C). Further, additional changes in angular  
203 velocity before and after stability indicate failed attempts or drastic posture changes  
204 in order to gain stability. During landings on wood, we found many small angular  
205 velocity peaks – the first  $4.8 \pm 4.0 \text{ rad s}^{-1}$  at approximately 50 milliseconds following  
206 touchdown, then  $7.4 \pm 4.2 \text{ rad s}^{-1}$  at 150 milliseconds, and again  $5.4 \pm 1.9 \text{ rad s}^{-1}$  at  
207 250 milliseconds, and finally at  $4.3 \pm 2.1 \text{ rad s}^{-1}$  at 410 milliseconds. During the final  
208 three angular velocity peaks, body angle relative to horizontal varied between  $52.9^\circ$   
209 and  $83.0^\circ$ . During landings on aluminum, small peaks in angular velocity were  
210 apparent as well, at  $7.6 \pm 2.4 \text{ rad s}^{-1}$  at 40 milliseconds after touchdown, again at  
211  $12.1 \pm 3.2 \text{ rad s}^{-1}$  at 130 milliseconds, and  $6.0 \pm 2.3 \text{ rad s}^{-1}$  at 330 milliseconds. During  
212 the final two angular velocity peaks, the body angle varied between  $63^\circ$  and  $84^\circ$ .

213         We found that there is a difference between time-to-stabilization between  
214 substrate treatments (Figure 4;  $F_{2,8} = 4.21$ ,  $p = 0.056$ ). On wood substrates, birds are  
215 stable at  $213 \pm 134 \text{ ms}$  following touch-down, on aluminum,  $151 \pm 99 \text{ ms}$ , while

216 stabilization on steel occurred at  $109 \pm 26$  ms following touch-down. The large  
217 standard deviations among substrate treatment indicate challenges of gaining  
218 control. In contrast, we found no significant differences among perch types for  
219 timing of leg extension prior to landing, which we assumed as an additional proxy  
220 for landing preparation (Figure 3D;  $F_{2,8} = 0.59$ ,  $p = 0.58$ ).

221 In addition to variation in control, we found that doves had stereotypical  
222 countermovements following landing, to aid in stability. Birds generally (74%) used  
223 their tail as a ventral counter weight (Figure 5A & 5B), flexing the tail ventrally  
224 under the perch to help counter the rotational motion of the body. In 26% of the  
225 trials, the birds used their tails in the opposite fashion, extending them dorsally over  
226 their heads and vertically above the perch (Figure 5A and 5C). This counter-weight  
227 function was observed most frequently on the wood perch (50% of the time), the  
228 most compliant substrate (Supplemental Video 2), followed by 17% of the time  
229 when landing on aluminum, and 10% of the time when landing on steel. Within  
230 each class of countermovement, the angle between the bird's body and tail did not  
231 differ significantly among substrates (tail-down:  $0.17 < p < 0.94$ , tail-up:  $0.78 < p < 0.98$ ;  
232 Figure 5A).

233

#### 234 *Perch Preference*

235 We found significant differences between perch diameters available in the  
236 habitat and the perch diameter upon which diamond doves were found in the wild  
237 (t-test  $p < 0.01$ ); Figure 6). Birds were found on perches averaging  $2.27 \pm 1.42$  cm in  
238 diameter, with a median value of 1.72 cm ( $n = 25$ ; Fig. 5A). The smallest perch found



239 in-use was 0.87 cm in diameter, and the largest was 5.4 cm. In contrast, the average  
240 perch available in the habitat was  $0.85 \pm 1.42$  cm in diameter. The minimum  
241 diameter observed was 0.06 cm in diameter, while the maximum was 8.54 cm. Birds  
242 were observed on average  $73.8 \pm 55.1$  cm from the nearest fulcrum within a shrub  
243 or tree.

244

## 245 **Discussion**

246       Variation in perch compliance impacts both take-off and landing abilities.  
247 Birds do not appear to capitalize on perch compliance during transitions, and as  
248 such, appear to suffer from inefficient transitions. Addressing our original  
249 questions, we found that take-off velocities are negatively impacted by perch  
250 compliance, while landing velocities remain unchanged. Additionally, we found  
251 minimal support for module coordination between the legs and wings during take-  
252 off. Instead, we found independent use of the modules, such that lower initial  
253 velocity due to the legs impacted later flight velocities, and during landing the wing  
254 and tail modules were used to aid in stabilization. Lastly, we found evidence for wild  
255 birds preferentially selecting perches of larger diameter, potentially minimizing  
256 effects of compliance during locomotor transitions.

257

### 258 *Compliance impacts velocity during take-off*

259       We found that take-off velocities are impacted negatively by perch  
260 compliance, while landing flight velocities are unaffected. During takeoff, birds lost  
261 energy to the substrate that they did not recover. We found no evidence that birds

262 modulate leg function to adjust for perch compliance; birds did not adjust the timing  
263 of force production (Figure 1B) or take-off force (Bonser, Norman & Rayner 1999).  
264 Thus, perches with higher compliance showed a lower instantaneous velocity at toe-  
265 off. Mathematical models of humans, bush-babies, and insects suggest that altering  
266 kinematic strategies on substrates of various compliance can dramatically change  
267 the resulting jump height (Alexander 1995). However, we saw no evidence for  
268 altering force-application timing during take-off in the doves. Both anole lizards  
269 (Gilman *et al.* 2012) and click-beetles (Ribak, Reingold & Weihs 2012) also do not  
270 alter force application to account for substrate variation.

271

#### 272 *Wing & leg modules function independently*

273 Wing and leg modules were found to not function within the same time  
274 interval, as previously predicted by Gatesy and Dial (1996). During take-off, the legs  
275 lose energy to the perch, causing initial low velocities. Following take-off, the wings  
276 increase flight velocity, but flight velocity by the third wingbeat remains inhibited by  
277 the initial lower velocities on compliant substrates. During landing, the wings  
278 decelerate birds to a consistent flight velocity before touch-down. However,  
279 following touchdown, both the wings and the tail (a third module) assist in  
280 counterbalancing the body to improve stability. Neurological or musculo-skeletal  
281 limitations offer a hypothesis to explain the staggering of module timing, as well as  
282 the inability to compensate for perch compliance. Transitions may be constrained  
283 by a neurological central pattern generator (Brown 1911; Kiehn 2006), wherein the  
284 timing between leg- and wing-modules cannot be adjusted. Further work exploring

285 the neurobiology behind modules is necessary to improve our understanding of  
286 potential neuromechanical limitations.

287         For birds, perfect force application during take-off via the legs may not be  
288 crucial, as the wings take over immediately following loss of contact with the ground  
289 (Provini *et al.* 2012). The dominant flight muscle, the pectoralis, produces the  
290 highest known power output within vertebrate locomotion (Schmidt-Nielsen 1972).  
291 Recent studies have established that the aerodynamic forces produced during  
292 energetically demanding slow flight (Tobalske *et al.* 2003) are modulated (Provini *et*  
293 *al.*, 2012, 2014; Jackson, Tobalske & Dial 2011) such that even during the most  
294 energetically demanding phase, power output can be adjusted depending on  
295 acceleration requirements. As such, it is arguable that the avian wing musculo-  
296 skeletal system is relatively 'overbuilt' in comparison to the leg, and so, is more  
297 capable of adjusting and compensating during transitions than the legs. While this is  
298 not observed in our data, we believe that had we extended the time interval over  
299 which we sampled flight velocity (i.e., not immediately following take-off), flight  
300 velocities would be indistinguishable between treatments.

301

302 *Birds show perch use preferences*

303         We found evidence that birds behaviorally adjusted for potential transitions  
304 by preferentially selecting larger-diameter perches that were further from the  
305 fulcrum compared with the distance to fulcrum in our laboratory tests (Figure 6).  
306 This may aid in compensating for inefficient transitions we have reported on highly  
307 compliant substrates (Figure 2, 3). Consistent with our assumption that stiffer

308 perches are better for high velocity take-off and better landing control, doves seem  
309 to prefer larger-diameter, stiffer branches in the wild. Further comparative work is  
310 warranted to examine the generalities of perch use in birds, alongside detailed  
311 habitat characteristic quantification (such as Gilman & Irschick, 2013).

312

313           Birds used in the laboratory study were raised in indoor aviaries with  
314 unlimited food supply and no predation pressure. Captive birds were not likely to be  
315 motivated to maximize performance in takeoff or landing, so caution is warranted  
316 when applying the results to an understand of transitions between substrates and  
317 the air in the wild. While the transitions we measured here did not optimize for  
318 variation in perch compliance, all birds completed their transitions without injury.  
319 During take-off, birds were able to accelerate in flight to equivalent speeds. During  
320 landing, birds all decelerated and reached stability successfully. As such, while the  
321 transitions were not efficient, they were effective. Further comparative studies are  
322 necessary to understand if inefficient transitions are a general character of birds,  
323 due to their powerful flight abilities.

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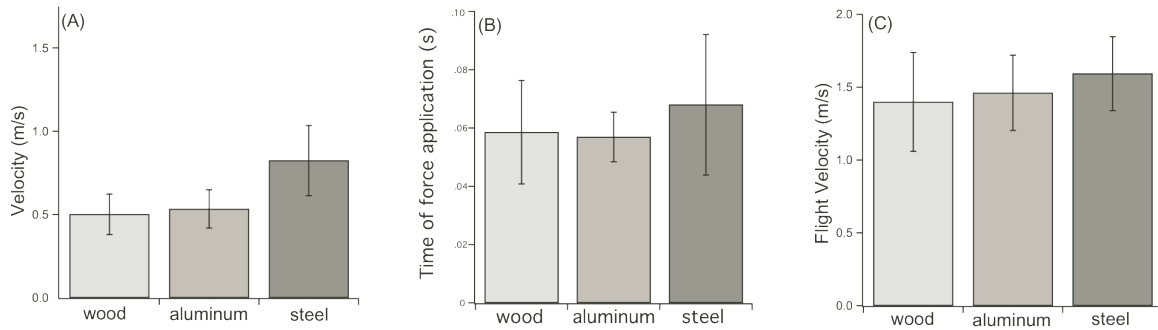
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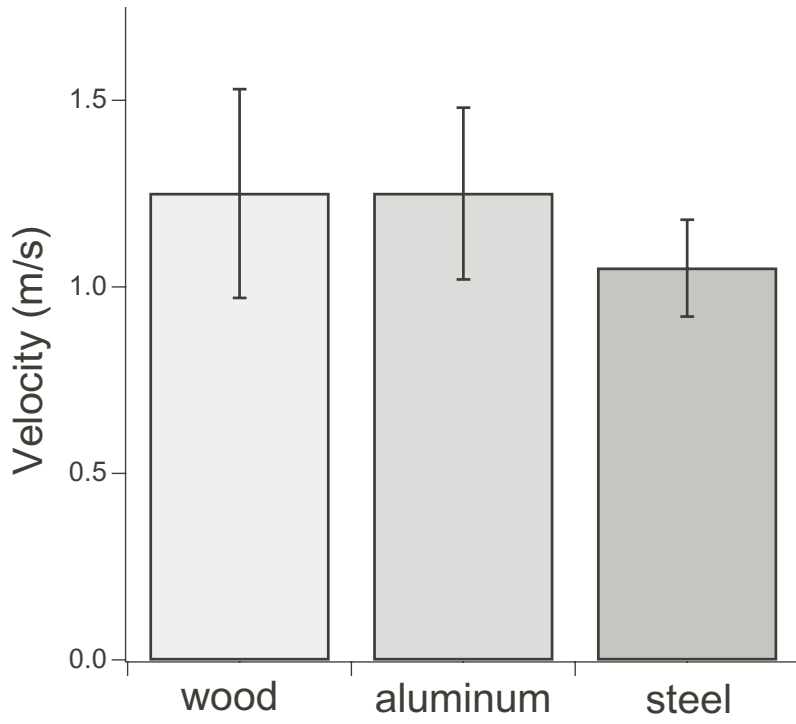
331 **FIGURES**  
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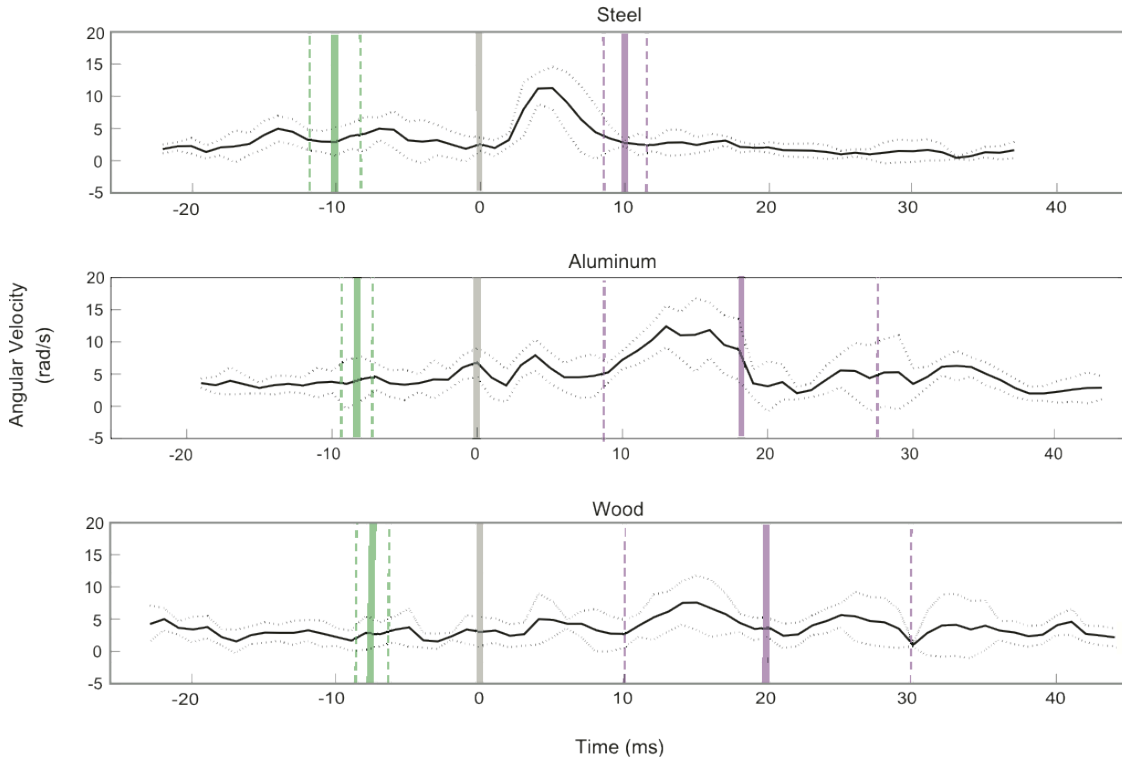
336 **Figure 1:** Take-off performance on perches of various compliance. Wood was the  
337 most compliant, steel, the least. (A) Resultant velocities at takeoff due to the legs (B)  
338 Timing of force application during take-off, (C) velocities at the third wingbeat of  
339 flight after toe-off.

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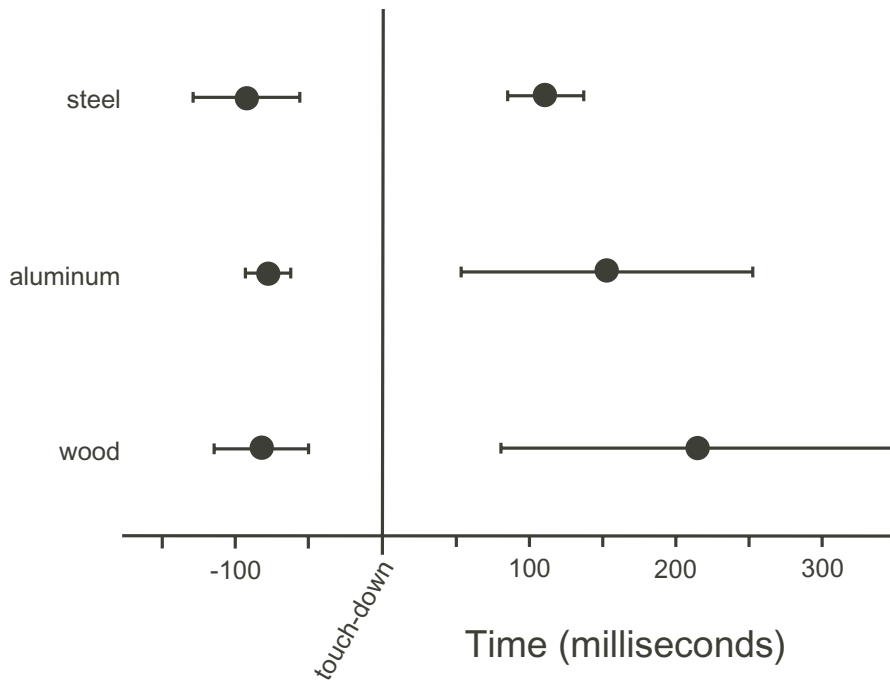
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**Figure 2:** Flight velocities of the bird during the last complete wingbeat prior to landing for all three perch types.



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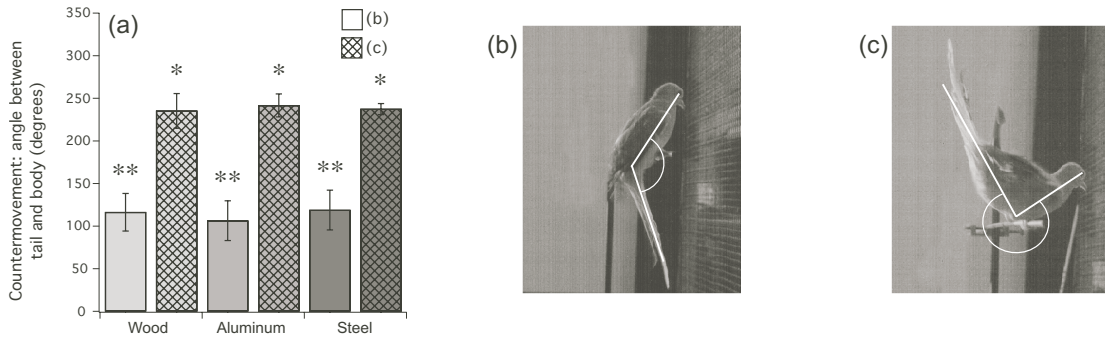
**Figure 3:** (A) Average angular velocities for ten trials of a representative individual (Dove 46) during landing on different substrates. Average timing (solid line) +/- standard deviation (dashed lines) is noted for foot extension (green), landing (gray), and the point the bird is considered stable (purple), following body reorientation. Instabilities are notable in following landing on the compliant substrates (aluminum and wood), indicated by additional angular velocities.



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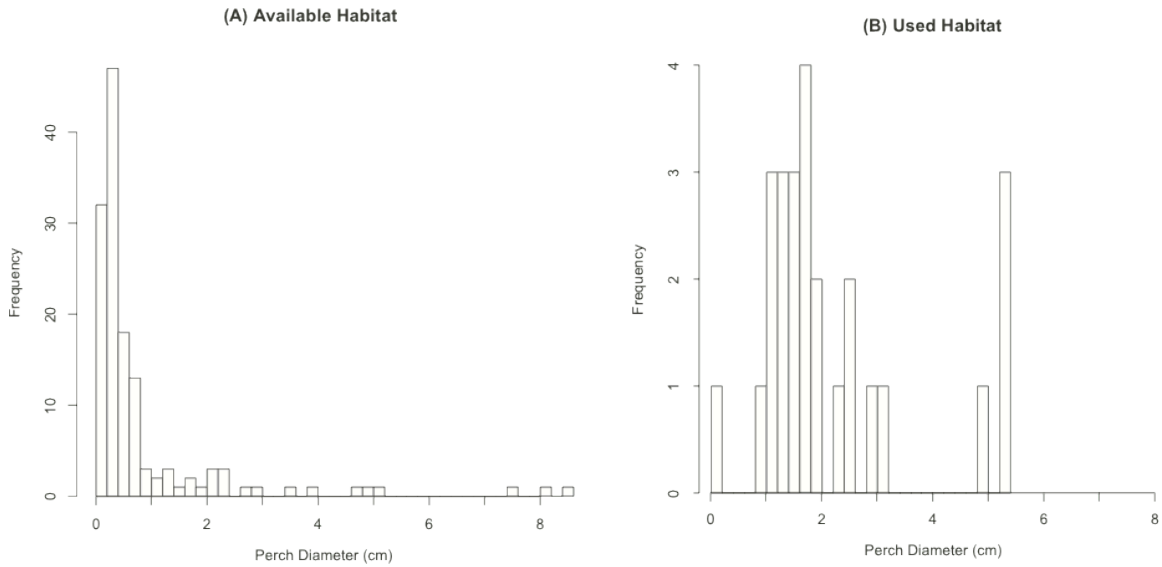
**Figure 4:** Timing of leg extension prior to landing (left) and point at which the bird has reoriented and is stable (right), relative to touchdown (time zero). Instabilities during landing are notable due to a longer time following touchdown and higher standard deviation until stabilization.





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**Figure 5:** (A) Angle of countermovements during landing to aid in stabilization for both countermovement conditions. (B) An example of a tail-under countermovement, reaching an average angle of  $113.7 \pm 23.6$  degrees. Between substrates, tail-under countermovements were statistically indistinguishable ( $0.17 < p < 0.94$ ) (C) An example of a tail-over countermovement, reaching an average angle of  $237.0 \pm 17.4$  degrees. Between substrates, tail-over countermovements were indistinguishable ( $0.78 < p < 0.98$ ).



419  
 420 **Figure 6:** Perch use in native Diamond doves. (A) Perch diameters available within  
 421 the habitat (B) Diameters of perches where doves were observed.  
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